



F-111



**22/12/1969:
Wing rupture during training
flight. Both pilots KIA with no
chance for ejecting**

Cause: Crack at the steel wing pivot fitting of F-111



Microphotograph of Crack Area . Flaw was not revealed during production NDT due to neighboring vertical elements masking the area

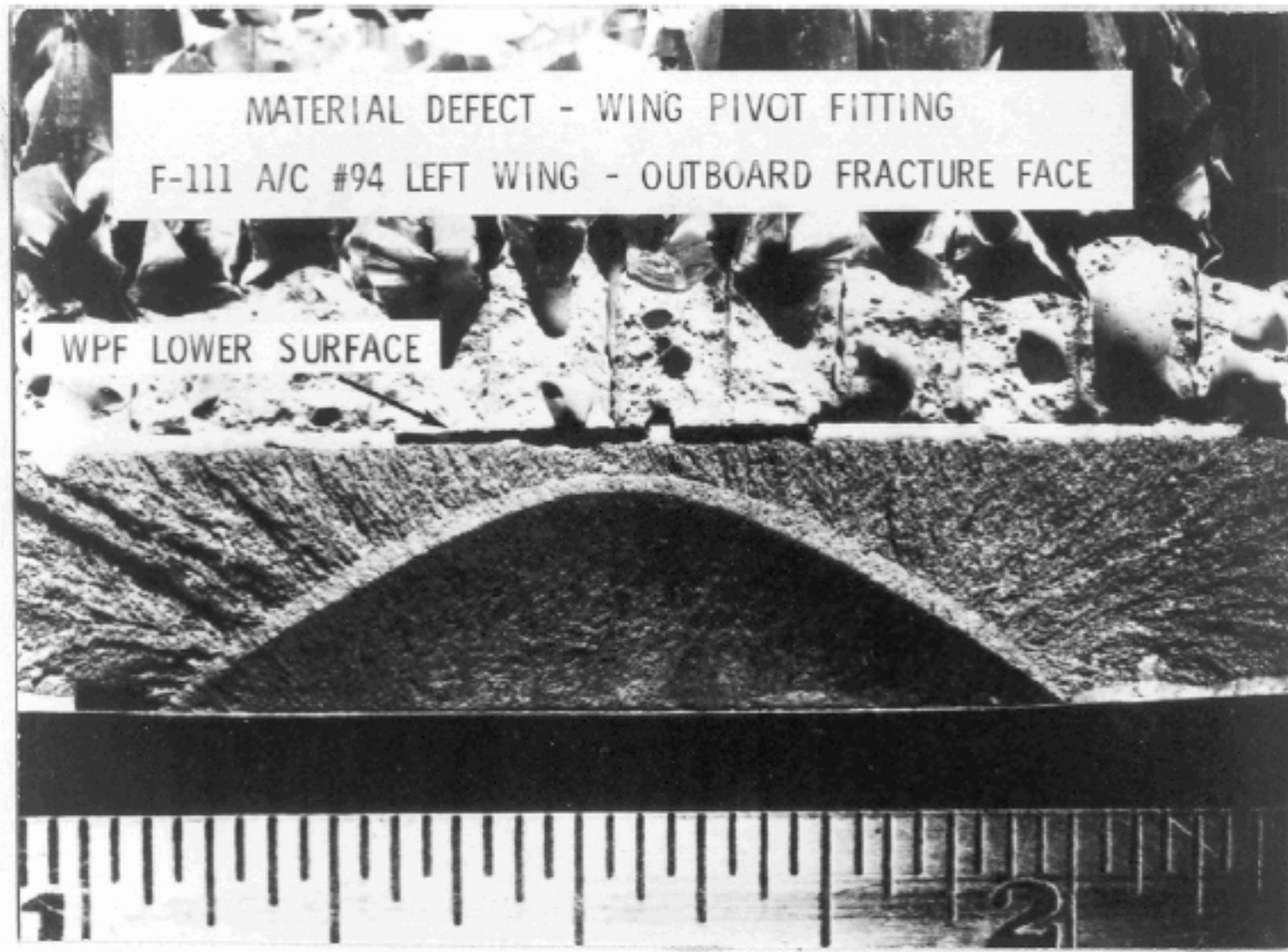
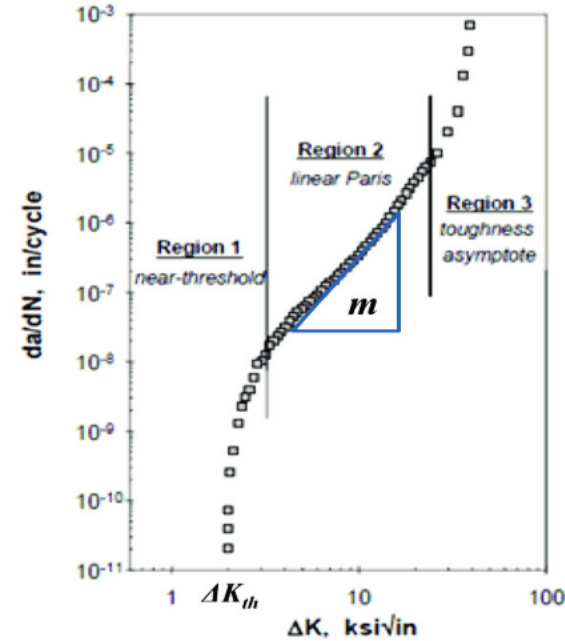


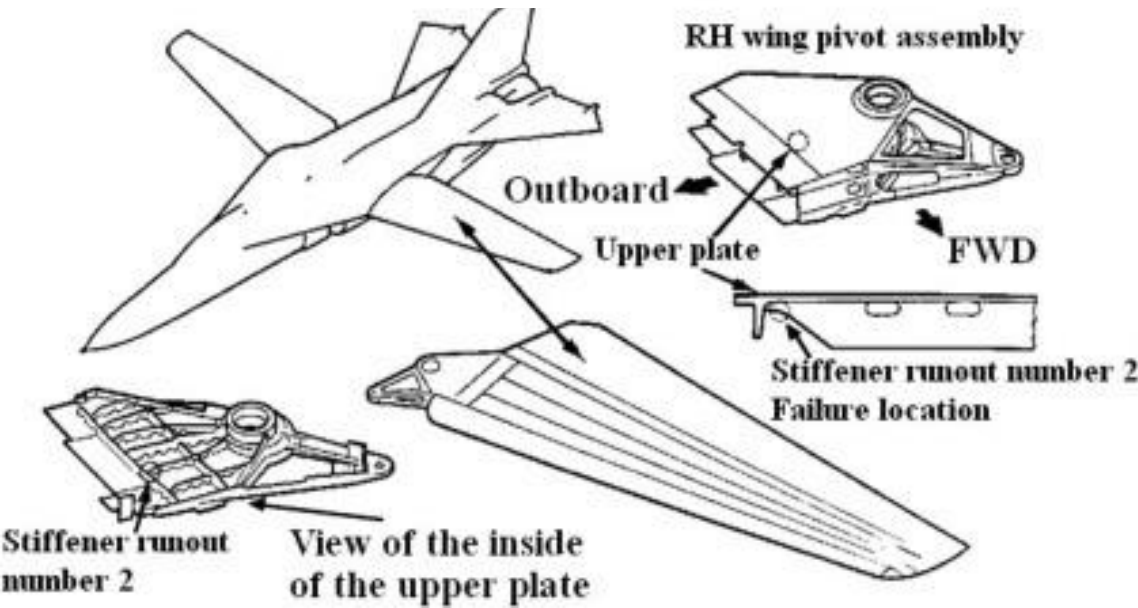
Figure 1.1.1. Origin of the F-111 Wing Defect [Rudd, et al., 1979]

Application of Boron Patches to carry structural loads, “deviating” the flaw area, in order to significantly reduce K

Commercial in confidence



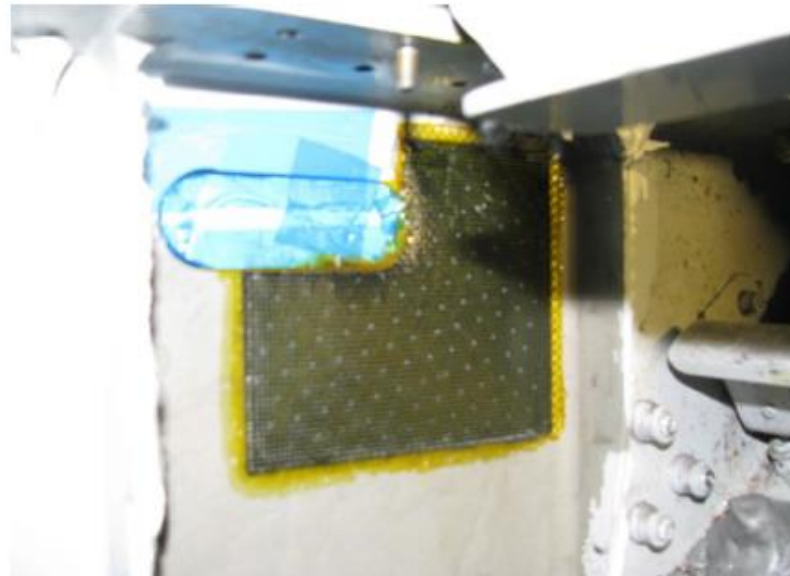
$$\frac{da}{dn} = C(\Delta K)^m,$$



Advanced composite repairs to metallic parts on ATR aircraft by GMI Aero



Repair of an **ATR-72**
aluminium floor beam
using a **bonded carbon**
patch, performed in situ
by GMI Aero, in
cooperation with the ATR
company.



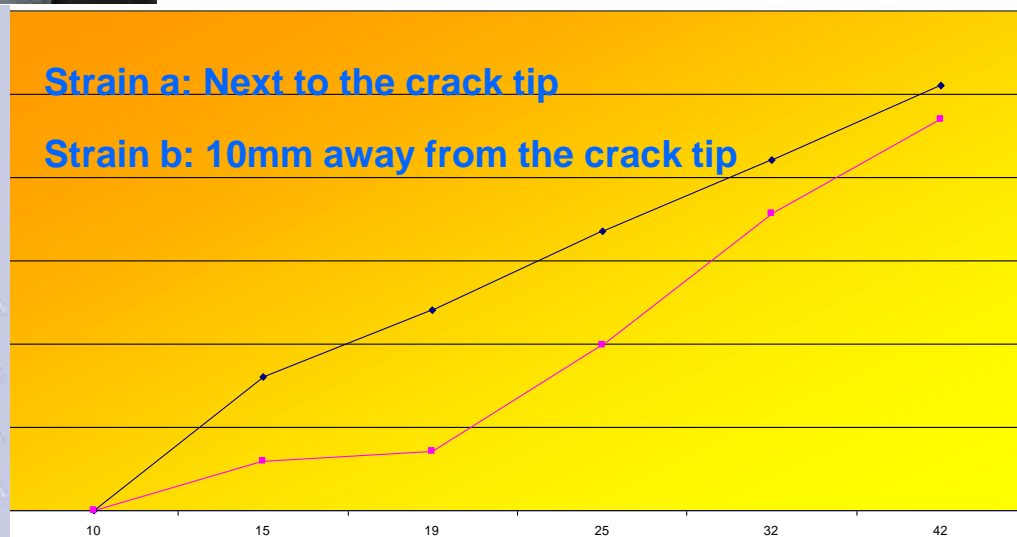


—◆— Strain a —◆— Strain b



Strain a: Next to the crack tip

Strain b: 10mm away from the crack tip





GMI
AERO

Commercial in confidence



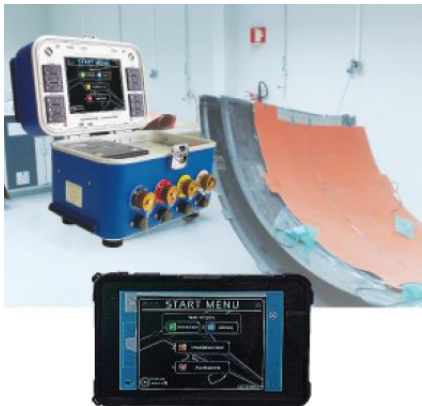
“Life will find a way”

- Michael Crichton, Jurassic Park

**“Bonded Composite Repair
will find a way”**

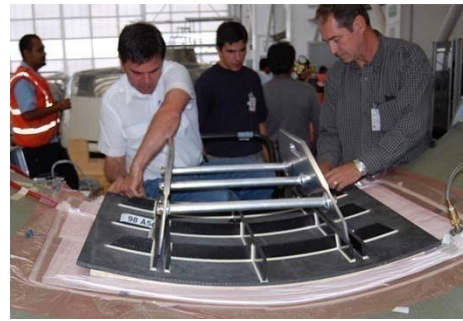
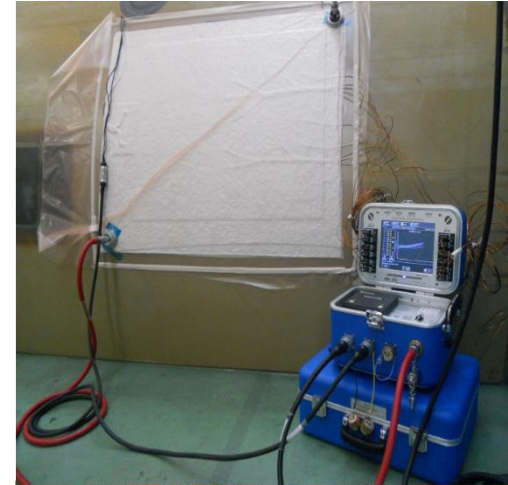
- Bremen E-LASS 29/1/2020

Bonded Composite Repair Methodologies – Latest Advancements and Developments



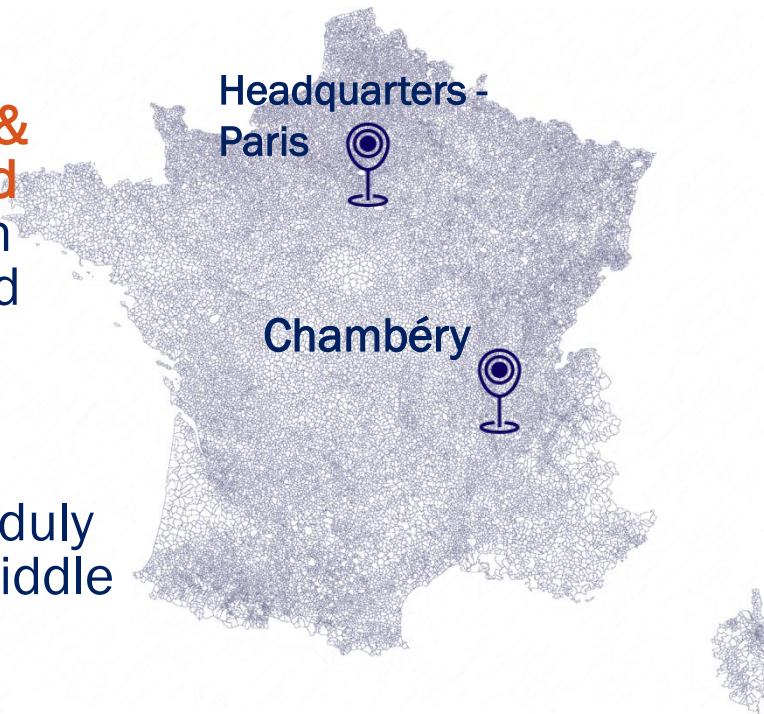
George Kanterakis
Sarah Baglione

- **Fabrication and Repair of Composite Structures** : Heating Systems, Process Models, Advanced Sensors, Advanced Bonding, NDT, Surface treatment.
- **Services** : Engineering solutions at customer's requests
- **Research** : for internal development & in Collaborative Projects with European Constructors and Universities
- **Consultancy & Training Worldwide** : at GMI's premises or at customer's facility



GMI Aero : The Company...

- More than **30 years experience**
- Team composed of **20 Engineers and Technicians**
- Departments of **R&I , Production, Design & Engineering** managed by **Researchers and Engineers** of long experience and in depth knowledge of all mechanical, sensors, and other physical aspects of this specialized science.
- **Represented in the world** through agents duly trained (Asia, Central & South America, Middle East, Europe).
- **Several agreements** of equipment through Airbus, ATR, Dassault, Bombardier, Boeing etc.
- **Qualified ISO 9001:2015**



Who are our customers ?

Aircraft manufacturers, MRO Centers, OEMs, Airlines, Universities



MoUs and Partnerships



- ✓ ADAMANT Composites, Greece
- ✓ Fraunhofer IFAM, Germany
- ✓ Hellenic Aerospace Industry, Greece
- ✓ Institute of Aviation, Poland
- ✓ IPSA, France
- ✓ Jordan University of Science and Technology, Jordan
- ✓ Malaysian Institute of Aerospace Technology, Malaysia
- ✓ Nat.Inst. for Aerospace Research "Elie Carafoli", Romania
- ✓ National Technical University of Athens, Greece
- ✓ SWEREA, SICOMP, Sweden
- ✓ University of Patras, Greece
- ✓ University of West Attica, Greece



Jordan University of
Science & Technology



National
Technical
University of
Athens



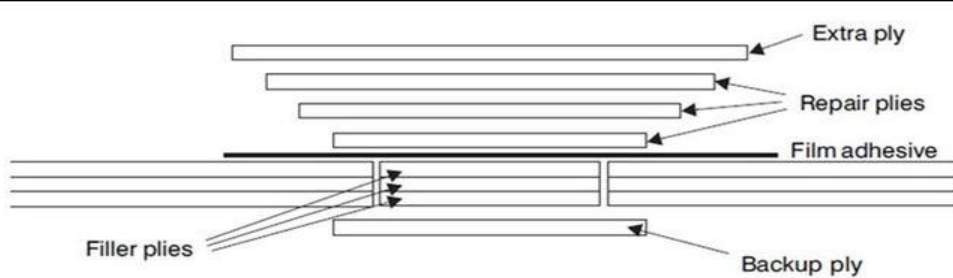
HELLENIC AEROSPACE INDUSTRY S.A.



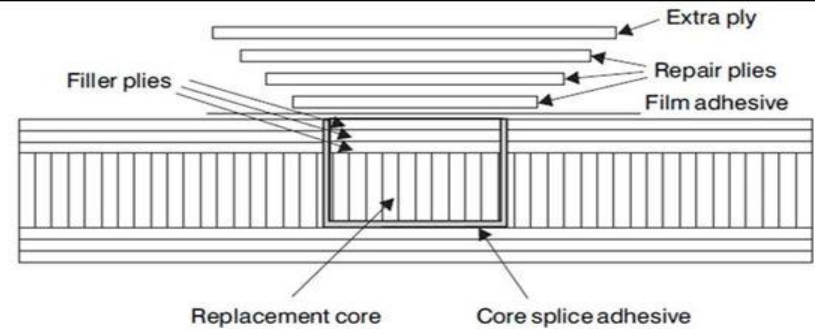
UniKL MIAT
Malaysian Institute of Aviation Technology



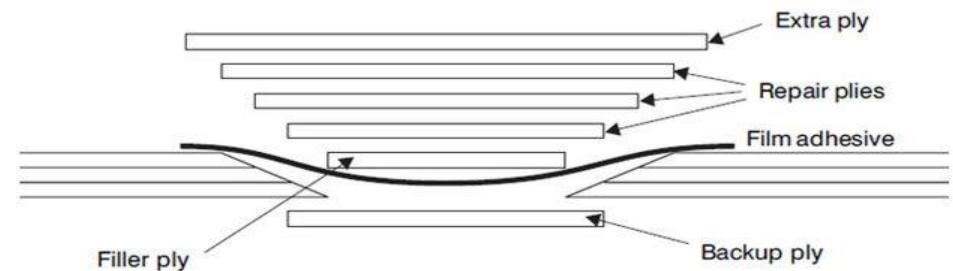
Typical Composite Repairs



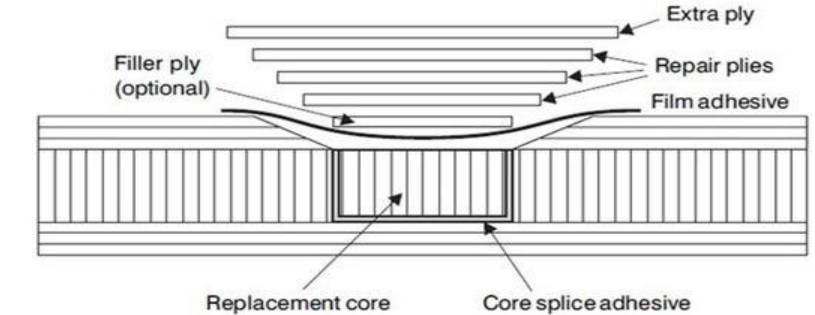
Patch repair



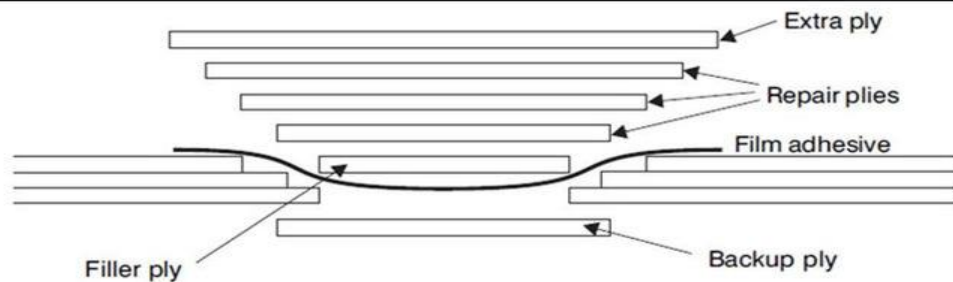
Patch repair



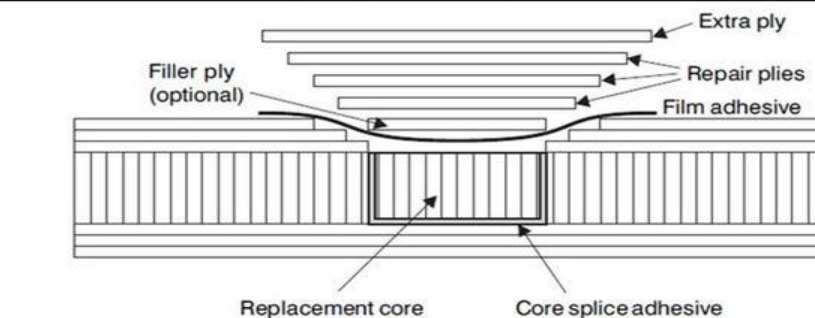
Scarf patch repair



Scarf patch repair



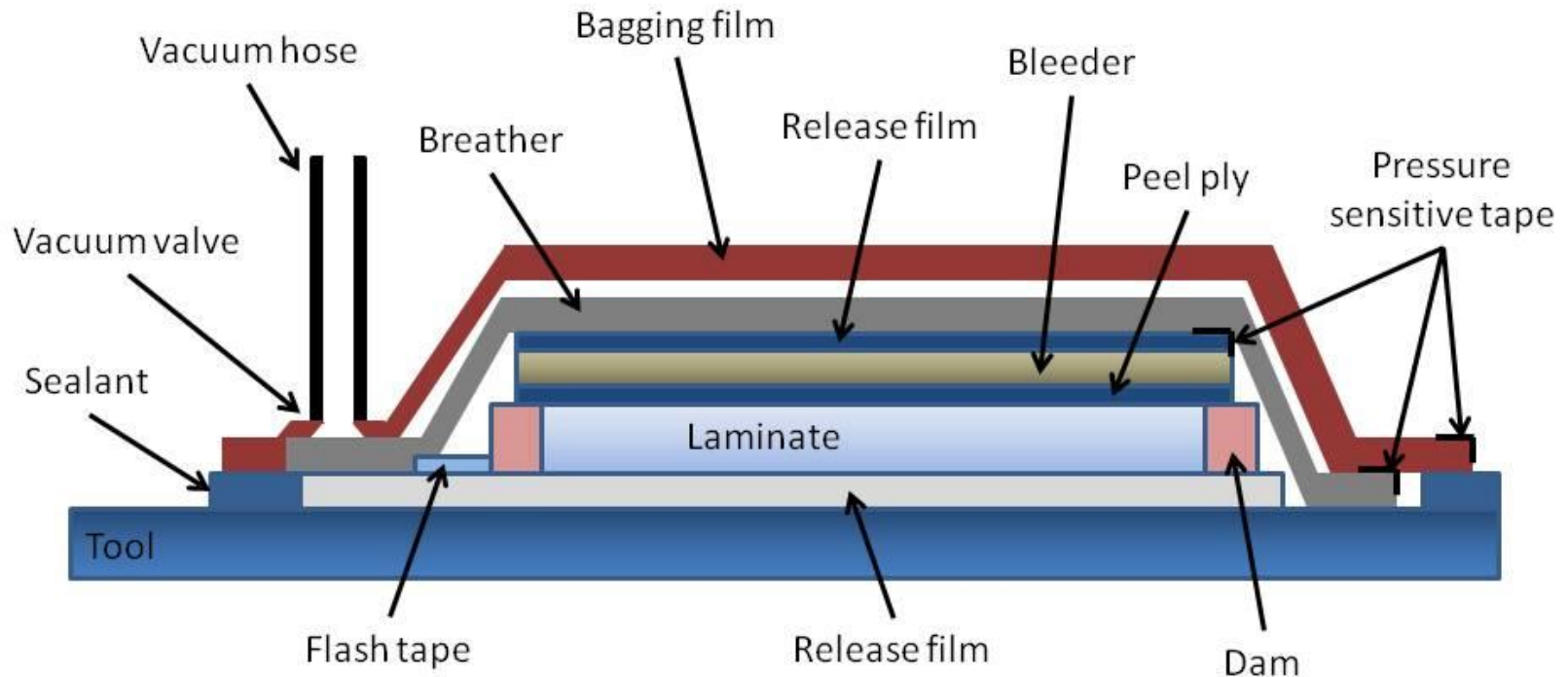
Stepped repair



Stepped repair

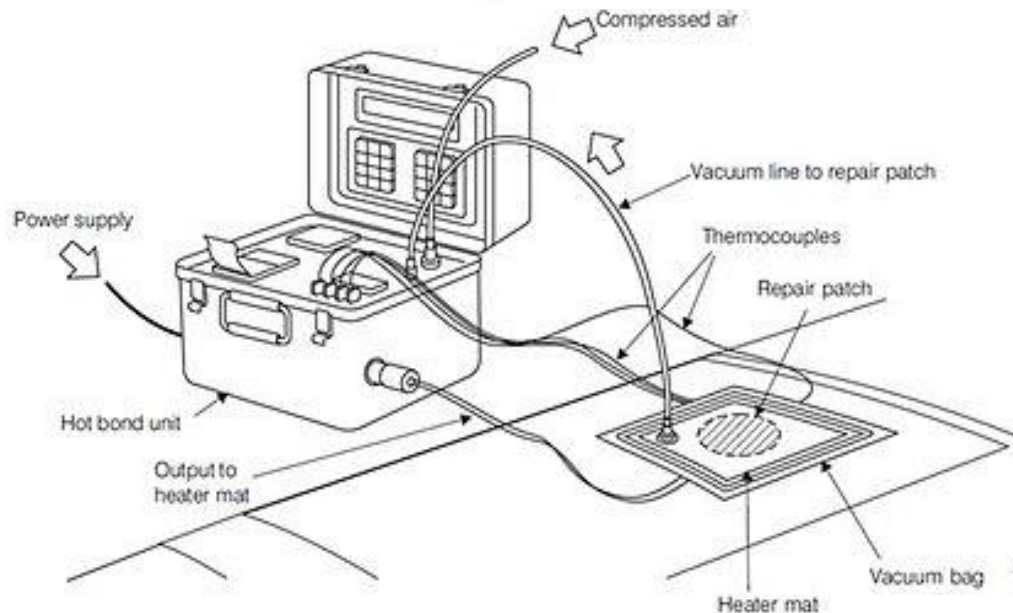
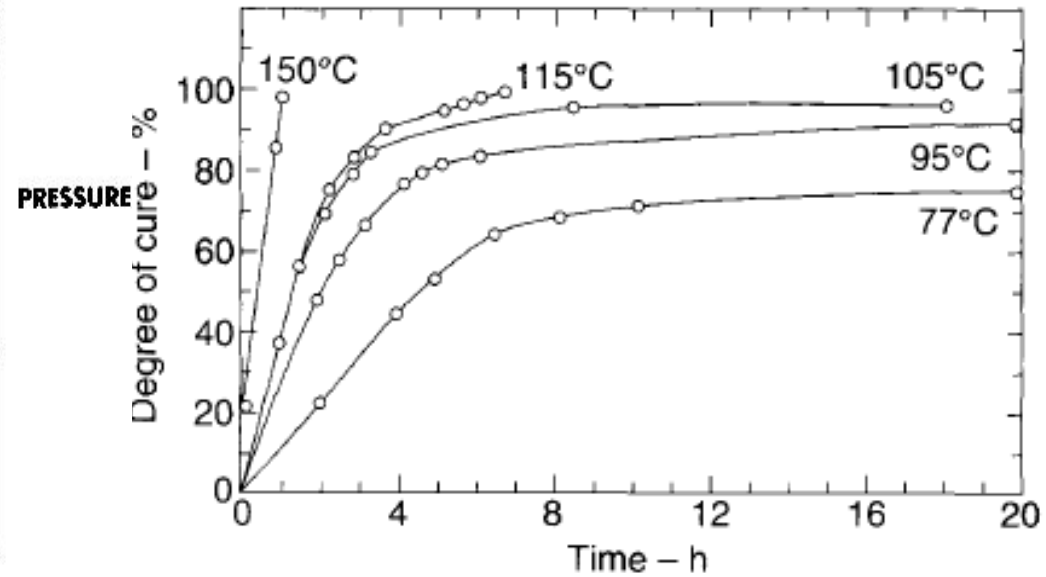
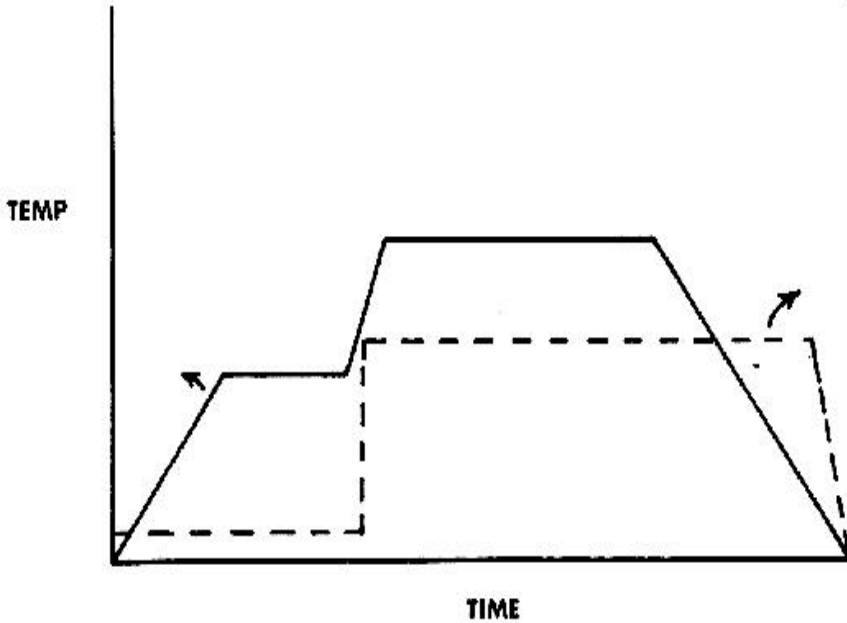
Vacuum Bagging

Vacuum bagging



The vacuum bagging technique can be used to improve the quality of composites produced by the wet lay-up method. A bagging film is placed around the laid-up composite material and is secured to the tool surface with sealant. Air is evacuated from the bag, leaving the composite under an external pressure of up to 1 atmosphere. This forces resin into any remaining voids and helps to ensure an even distribution. Higher viscosity resins can be used in comparison to the wet lay-up technique.

Repair Curing



whole,

ANITA Family of Bonding Consoles



Anita EZ



Anita EZH



Anita QS



Anita EZ4Z



April (PR)



Anibolt



Anita 6Z


Wide range of heating elements

Heating Technology



FGH and SXH High Temperature Composite Curing Blankets

Product Highlights

- ✓ Designed for use with the newer high temperature thermoplastic and polyimide composite materials
- ✓ Highly flexible up to a 1" (25mm) radius
- ✓ Compatible with ACR® hot bonders and your current equipment
- ✓  CE RoHS Compliant

Specifications:

- Heating element and a 1" (25mm) layer of high-density fiberglass is covered in an abrasion resistant fiberglass cloth (FGH) or Samox® cloth (SXH series)
- Maximum exposure temperature:
 - FGH series: 800°F (425°C)
 - SXH series: 1100°F (593°C)
- Power density:
 - FGH series: 7 watts/in² (0.011 watts/mm²)
 - SXH series: 13 watts/in² (0.020 watts/mm²)
- Dielectric strength of over 2000 volts
- Power cord 6-foot (1.8m) long with choice of power plug

Temperatures up to



1100°F (593°C)



Surface Preparation – Innovative toolings have been designed to allow technicians to perform easily typical works on carbon surface for patch installation



Stepping

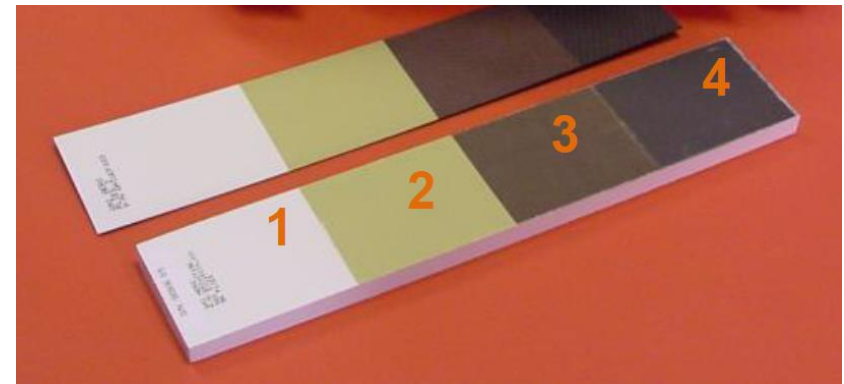
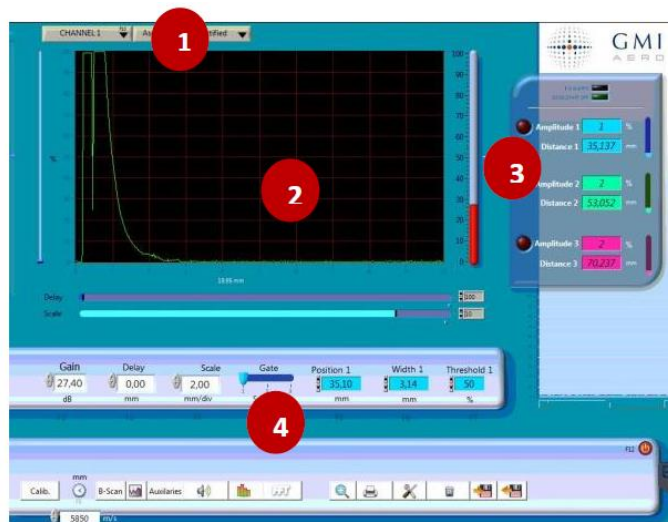


Scarfig



A complete set of tools for all tasks – Easy to handle and use – Presented in a complete Mobile Workshop

Elisa U/S NDT Console

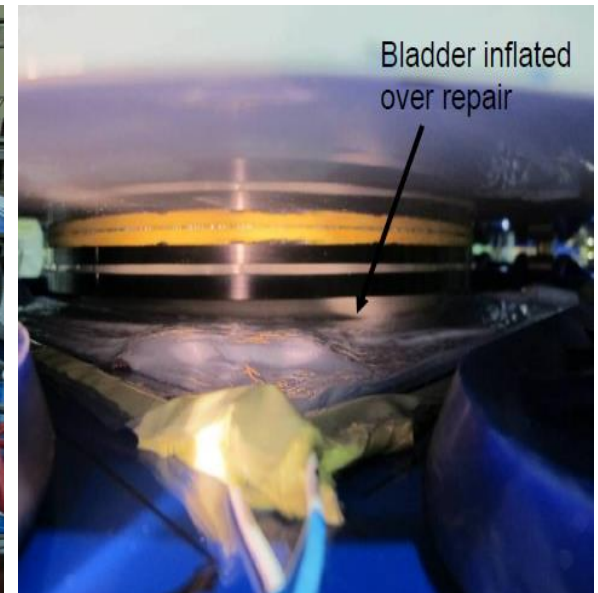
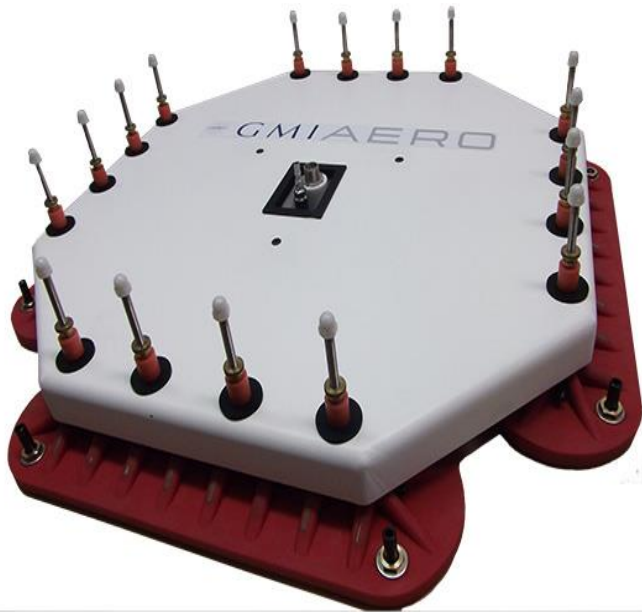


Range of Calibration Specimens

Advanced Software Tailored to
Composite Repairs

OLGA Positive Pressure Application

Repairs using OLGA patch preparation (to overcome porosity issues) and secondary bonding on aircraft

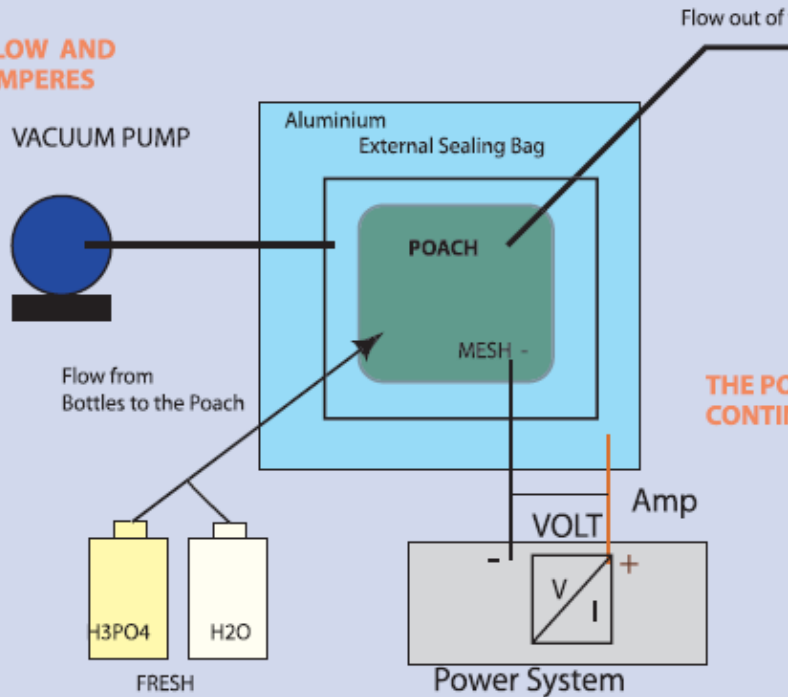


GILDA Phosphoric Anodizing



AUTOMATIC ADJUSTMENT OF

- FLOW AND
- AMPERES



Autoclaves & Ovens



*Various dimensions and specifications
Fully controlled by our computers
GMI ANIFIB*



*Model standard of internal dimensions
2,5 x 2,5 x 4 m
Fully equipped.
With vacuum system
With **GMI ANIFIB** Computer control*

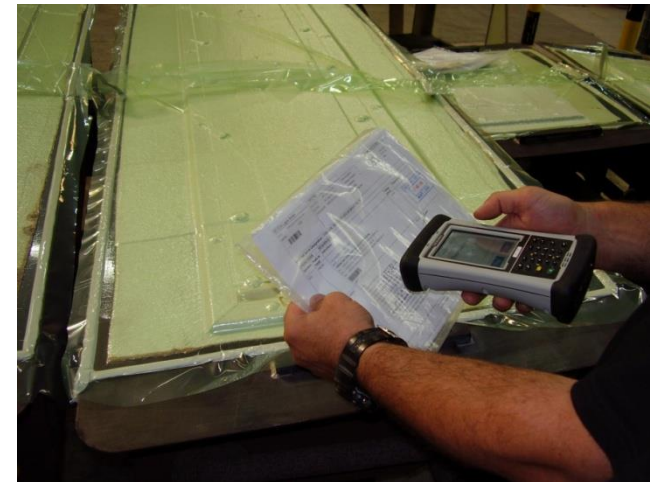
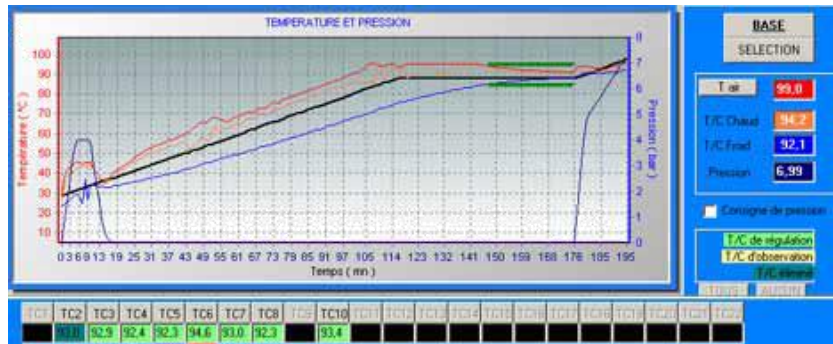
GMI autoclaves are controlled by our system **ANIFIB™** that offers several unique solutions:

The **ANIFIB™** system is composed of two sub-systems:

- An Industrial Computer called itself ANIFIB and which is mounted near the oven/autoclave.
- A PC linked to the **ANIFIB™** computer as a supervisory station with cycle programming and supervisory software.



Statut	Temperatures de Chauffage / Refroidissement	Pression (bar)	Temps (min)	Temps de détente	Temps de attente	Temps de cycle
EN CYCLE	100.0 °C 131.5 °C	2.2 bar 2.20 bar	195 min 43 min			545 min
EN CYCLE	64.1 °C 73.5 °C	0.0 bar 0.00 bar	59 min 178 min			104 min
EN CYCLE	177.2 °C 175.5 °C		78 min 332 min			78 min
FIN DE CYCLE	- °C					
EN CYCLE	149.9 °C 156.7 °C	2.2 bar 2.22 bar	165 min 22 min			28 min
FIN DE CYCLE	- °C	- bar				



Example of Autoclaves Installation

With Anifib, GMI has developed an activity of autoclave control.

Helping companies ease their production process and qualify it.

We are now turning to a development of **Industry 4.0**



Research & Innovation

Participation in **more than 30 EU Research Programs**
focusing on bonded composite repairs

FP5, FP6, FP7, **H2020**, Cleansky, Cleansky 2,
Research for the Benefit of SMEs, Eurostars, EUROGIA+,
Ten (10) Cleansky & Cleansky 2 R&D Projects

In cooperation with :

Co-funded by:

AIRBUS

 **LEONARDO**



 **eurostars™**

 **DASSAULT**
AVIATION

 **SAFRAN**
AIRCRAFT ENGINES

ATR

 **EADS**

EUROGIA +



A EUREKA initiative



 **Clean Sky2**

GMI R&D Activities 1997 – 2020

58 Research Papers

“Revolutionizing Aircraft Materials and Processes”, Edited by Sp.Pantelakis and K.Tserpes
Chapter: “Bonded Repair of Composite Structures”

G. Kanterakis, R. Chemama, K. Kitsianos

H2020: MG.1.1+2014.-RIA 636494-2

Quality assurance concepts for adhesive bonding of aircraft composite structures by advanced NDT – ComBoNDT

Consortium: Fraunhofer IFAM, EADS, CNRS, UoP, ASTRIUM, EASN



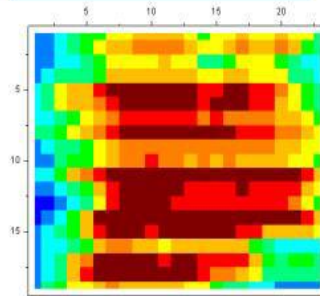
State of the art



ENDT techniques can detect:

- **Single surface contaminations** (pre-bond) on **simple** sample **geometries**
- *Weak bonds* due to single contaminations/poorly cured adhesive (post-bond) on simple geometries

1st step



Maturation of ENDT techniques to detect:

- Different **multiple surface contaminations** down to a certain threshold value (pre-bond) on **test coupons**
- *Weak bonds* due to multiple contaminations/poorly cured adhesive (post-bond) on test coupons

2nd step



Adaptation and improvement of ENDT in terms of:

- *Pre-bond/post-bond* inspection on pilot samples with **realistic geometries**
- **Automation and industrialization** of ENDT techniques, including its use on a demonstrator and automated data evaluation
- **Validation** of measuring results/round robin test

Final results/innovations of ComBoNDT



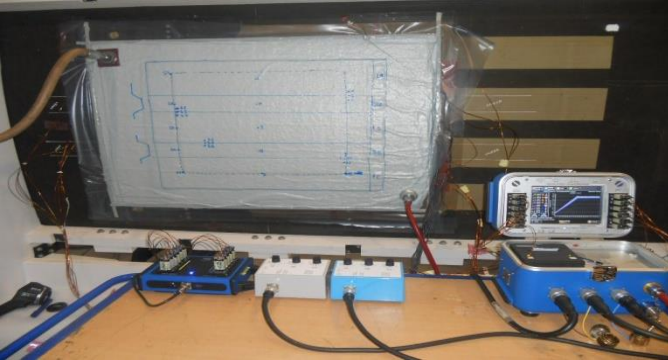
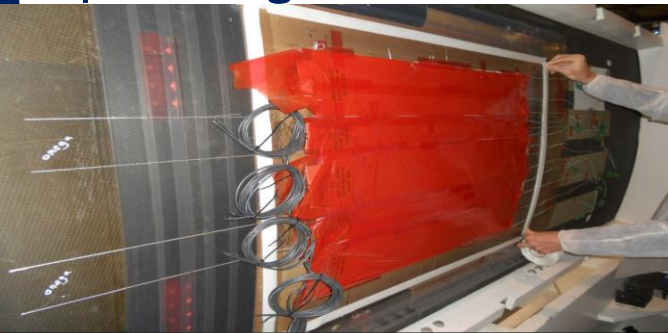
Validated ENDT techniques for:

- Surface quality assurance (pre-bond)
 - Adhesive bondline quality assurance (post-bond)
- ...for integration into future adhesive bonding processes

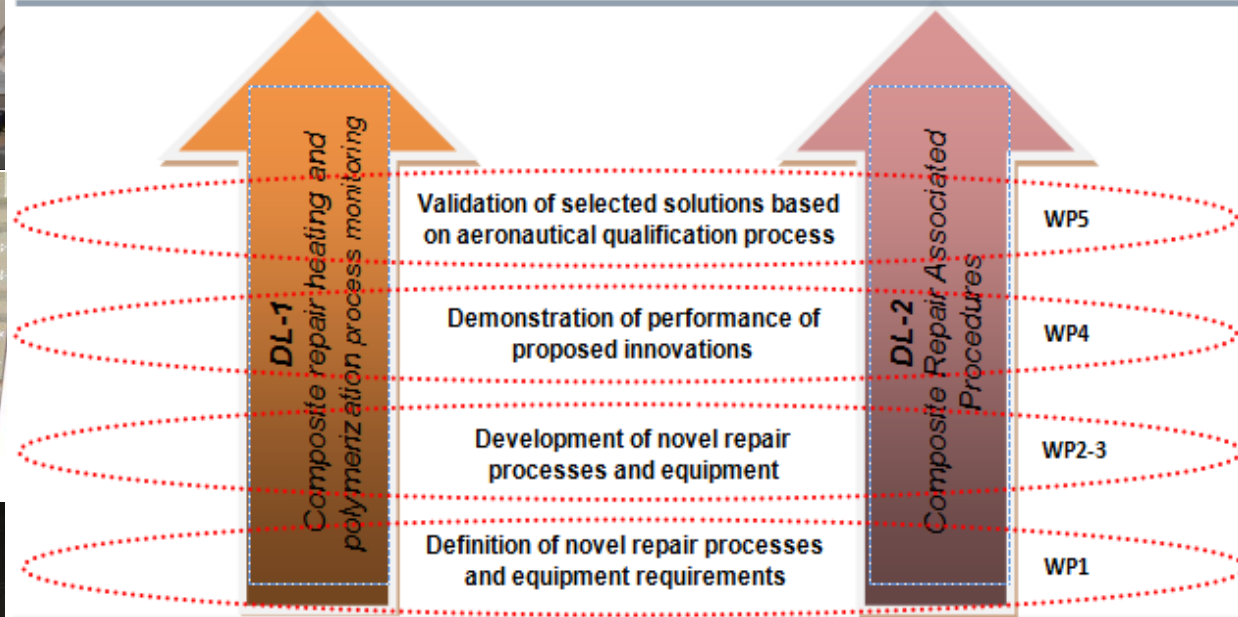
Cleansky 2: AIR-02-05-686701

Novel Processes & Equipment in Composite Repair Technology – NEWCORT

Topic Manager : Airbus



Support of mid-term needs for innovative faster repair process of monolithic composite airframe and long-term repair processes for future epoxy and thermoplastic materials



NEWCORT Background knowledge in Bonded Composite Repair Processes

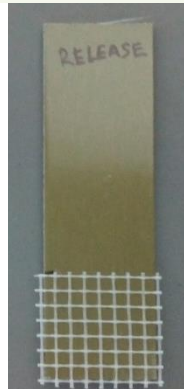
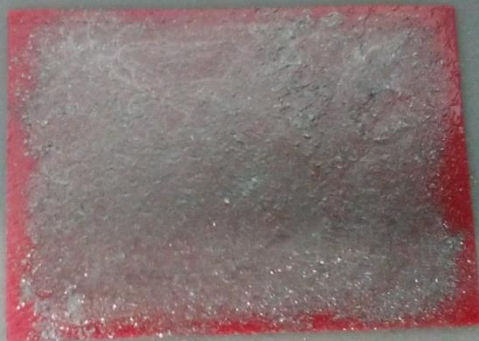
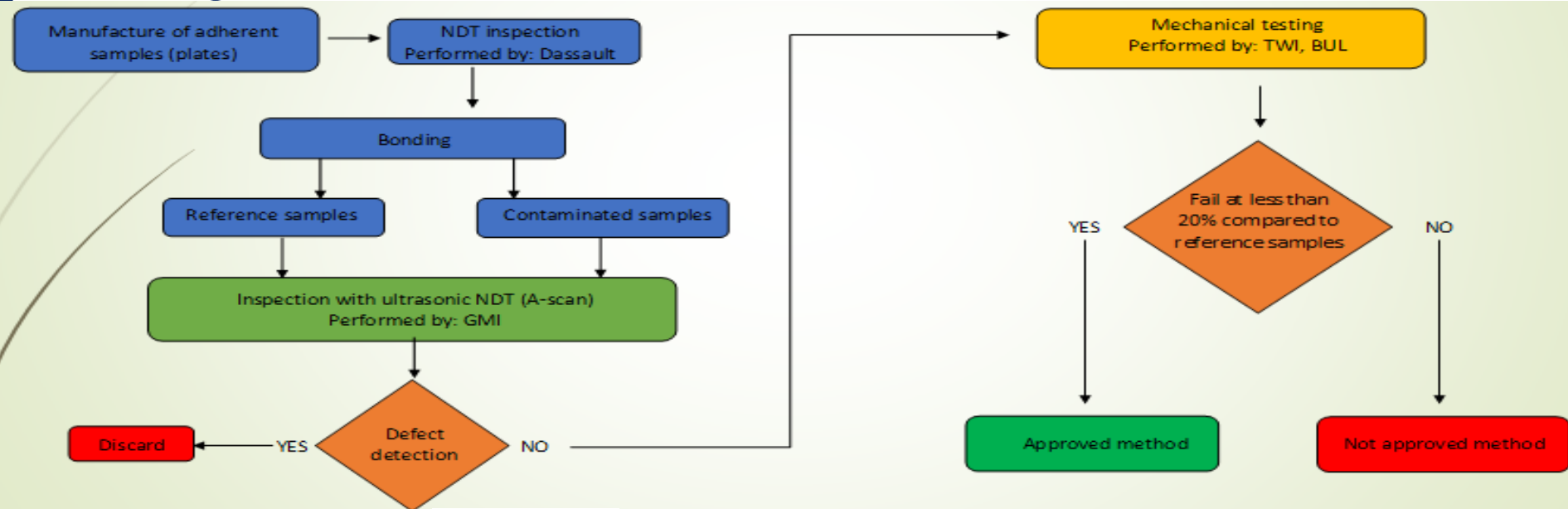
GMI commercially available Composite Repair Equipment and Solutions
(110-page equipment catalogue)

NEWCORT Partners' R&D projects in Bonded Composite Repairs
(~20 R&D projects)

Cleansky 2: AIR-03-02-831882

Non-destructive testing (NDT) of bonded assemblies–
SealedWithoutAKiss

Topic Manager : Dassault Aviation



Latest innovations related to bonded composite repair application

Participation to EU Research Programs Brite Euram (FP5), FP6, FP7, H2020, Cleansky JTI, Research for the Benefit of SMEs, Eurostars & EUROGIA+

	Innovation Description	Challenges answered
1	Repairs using OLGA (positive pressure application equipment) for patch preparation and secondary bonding on aircraft.	Enhancement of bonding quality, and overcoming patch porosity issues
2	Adaptive (multi-sectorial) heating blankets using existing 2-zone ANITA bonders combined with variable insulation.	Improve temperature homogeneity on geometrically complex structures.
3	“Plug-and-play” 24-Thermocouple scanner , together with appropriately prepared sensing mats.	Increased quality control requirements during repair on complex structures.
4	Real-time transmission of repair data, for curing duplication using dielectric sensors to enable curing degree monitoring.	Increased quality control requirements during repair on complex structures.
5	Heating solutions for quick repairs of small number of plies and limited dimensions tailored to aircraft needs.	Fast aircraft turnaround when limited repairs need to be applied.
6	CONDUCTOR consumable heating blankets that could be cut in different shapes to adapt to geometrical requirements.	Improvement of T homogeneity and suppression of blankets' lead time.
7	MAGNASENSE magnetostrictive sensors , for fast and accurate strain mapping of new or repaired composite structures.	Ensuring bonding adequacy both for SHM and certification requirements.
8	Autonomous mobile composite repair workshop to be used by airlines that need to perform repairs outside of their hangars.	Flexibility in repair application. Reduction of aircraft turnaround time.
9	Advanced heating system & control mode for homogeneous Out-Of-Autoclave curing of large composite repairs - ADVANCED	Reduction of energy consumption and CO ₂ footprint for repair / production.
10	Heating Bolts provide localized heating to repair damages associated with drilling & reaming operations in carbon laminate.	Curing of resin inside drilled holes, achieving required T homogeneity.

In cooperation with



Co-funded by:



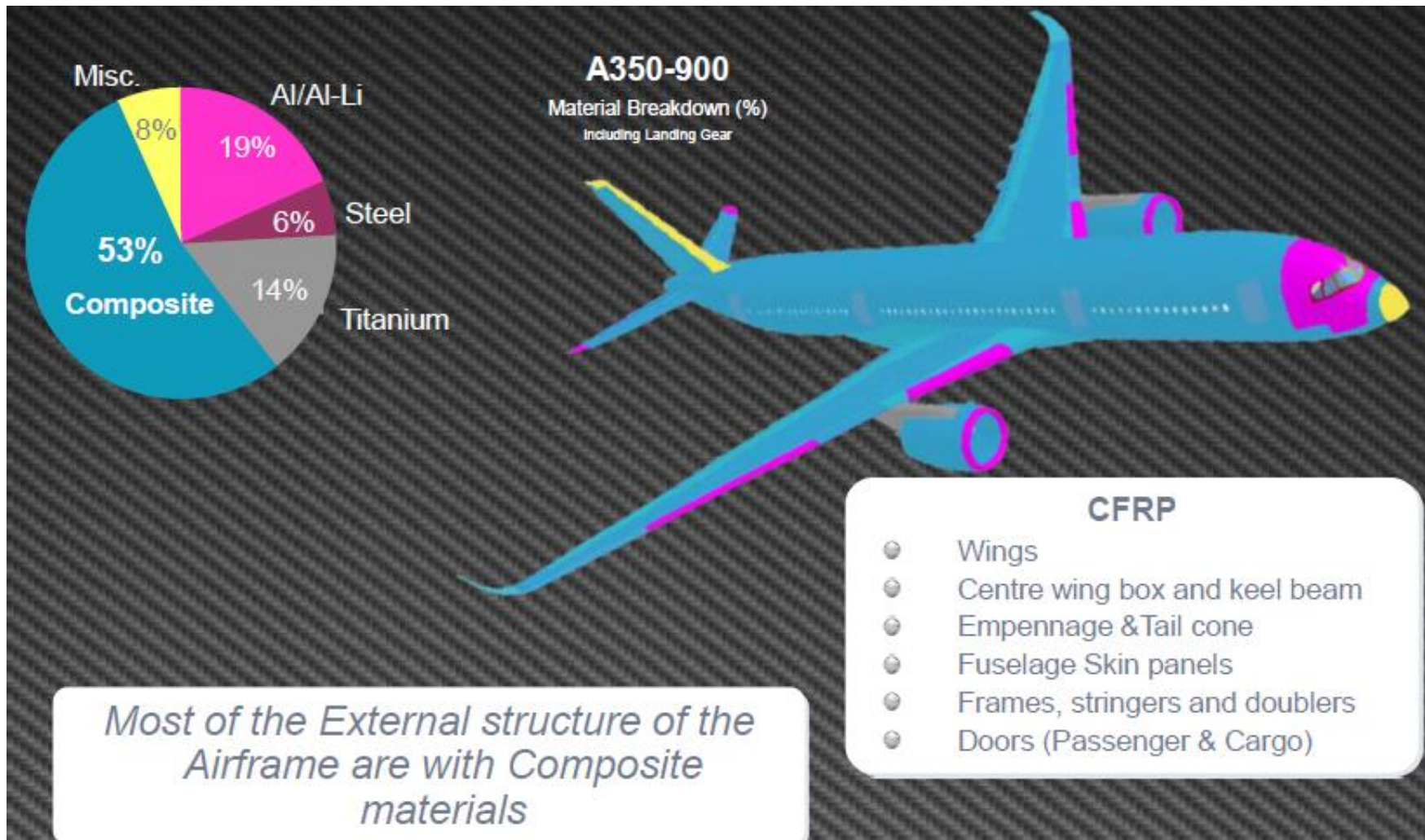


GMI
AERO

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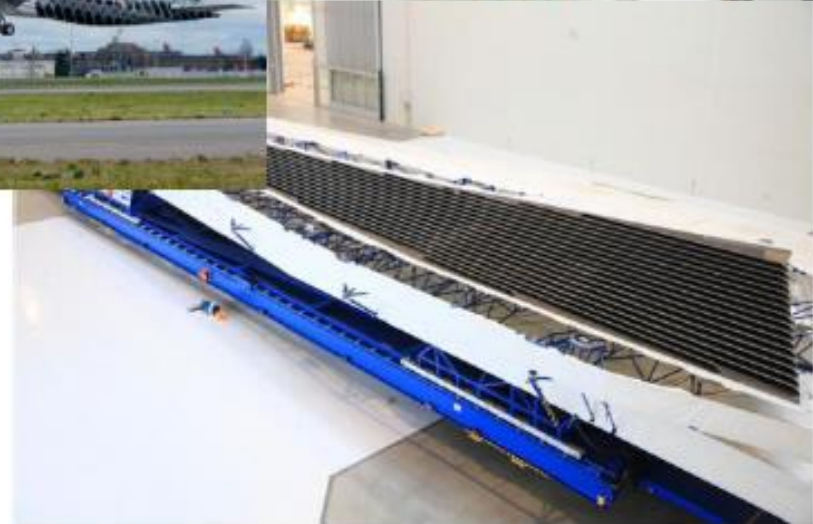
A350 : Composite Materials overview

(Source Airbus)

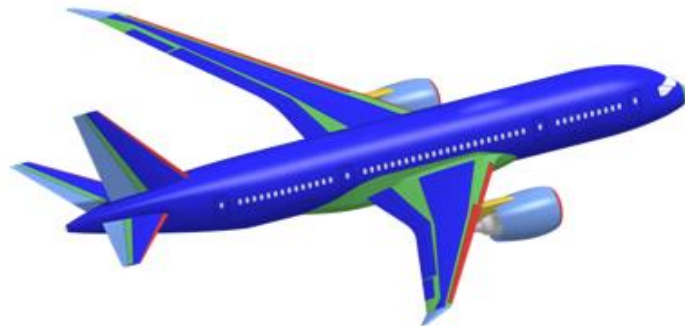


A350 : Involving Large & Complex Structural Panels

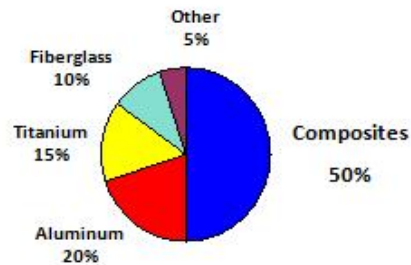
(Source Airbus)



Increase of volume of Composites per aircraft



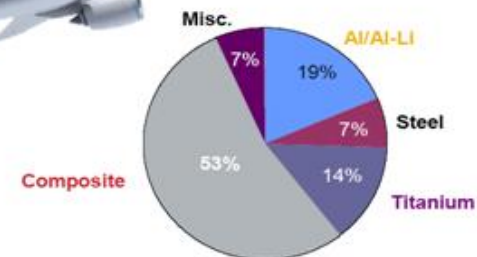
B-787



A350-900 XWB – Material breakdown



A350-900 XWB
(%) Including Landing Gear



Together with
Increase of complexity of Composite parts

Answer to Challenges:

- a. Adaptive Heating Solutions**
- b. Integration in 4.0 concept**
- c. Training & Continuous Professional Development**

Adaptive heating solutions for aeronautical applications

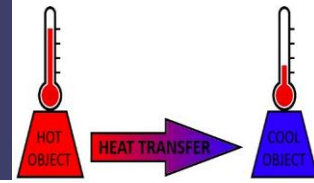
If the only tool you have
is a hammer, everything
looks like a nail.

To change your perspective
(and find new solutions.)

Pick up new tools.



Please consider...



Carbon Structures Hot Bonding Issues

- The main challenge : to match in temperature the resin manufacturer specs: respect of a set-point of temp. at +/-5 °C max. on the whole surface.
- Heat transfer on structures heated by conduction through mats leads to **gradients up to 40 °C** and more, due to :
 - Large dimensions
 - Orientation of structures
 - Variation of skin thicknesses
 - Variability of materials covered by the patch

Question!



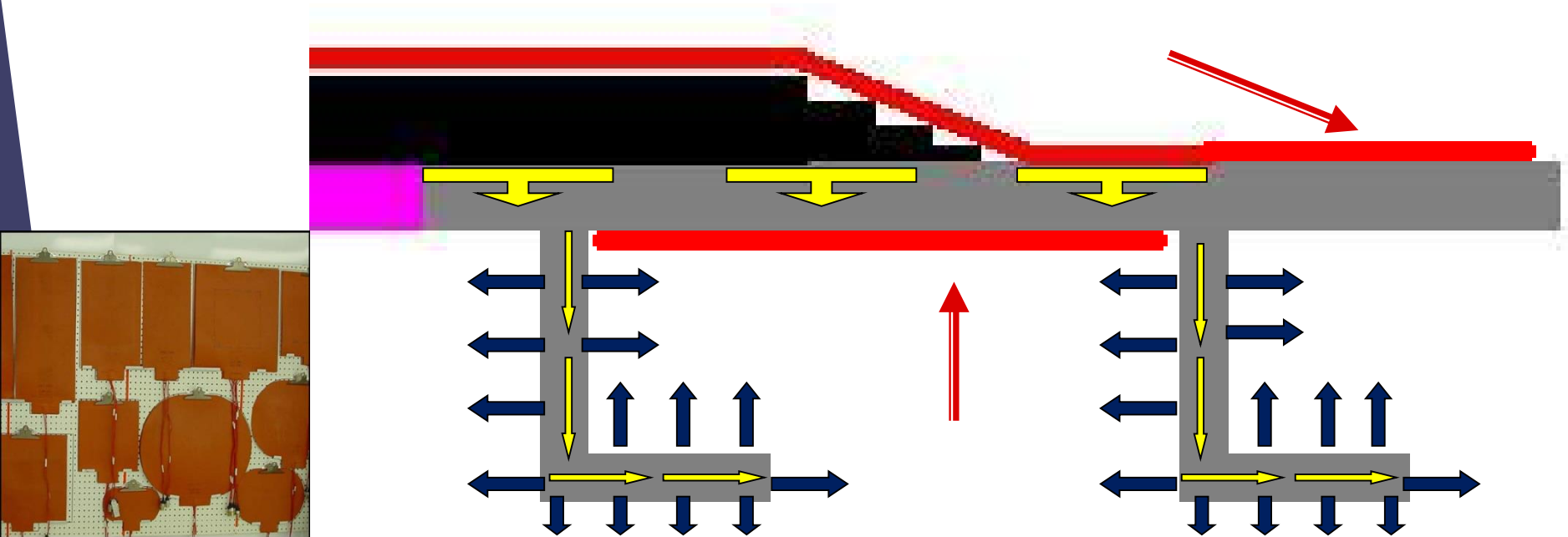
Question: Why do we have temperature differences?

Answer: The problem starts due to the continues interaction of the thermal transfer modes, as follows:

- Heat is **directed to the repair area** (composite patch & peripheral structure) **through the heating** blanket by conduction, at a uniform rate across the blanket surface (increase of temperature).
- Heat is **propagating through conduction to the whole structure** which is surrounding the repair according to its geometry (i.e. in a non-uniform way) and, at the same time,
- Heat is **lost by the whole repair area through convection to the environment** according to the geometry of free surfaces of the surrounding structure (i.e. in a non-uniform way)

As is it well understood the result of uniform thermal heating, minus non uniform heat losses due to conduction and convection
is a non-uniform temperature distribution

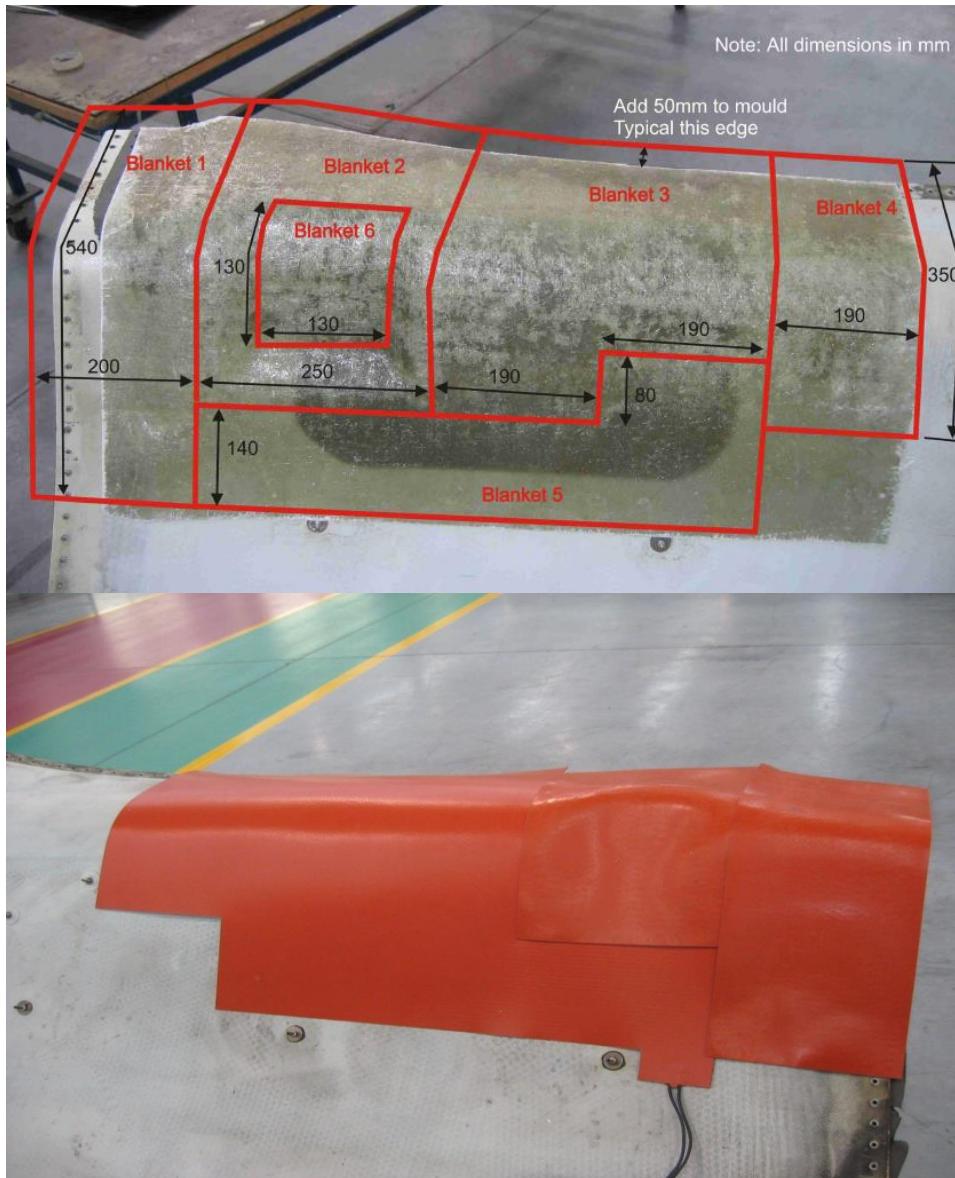
Tips for achieving better temperature distribution



Tip: Impose additional heat to areas affected by “heat sinks”

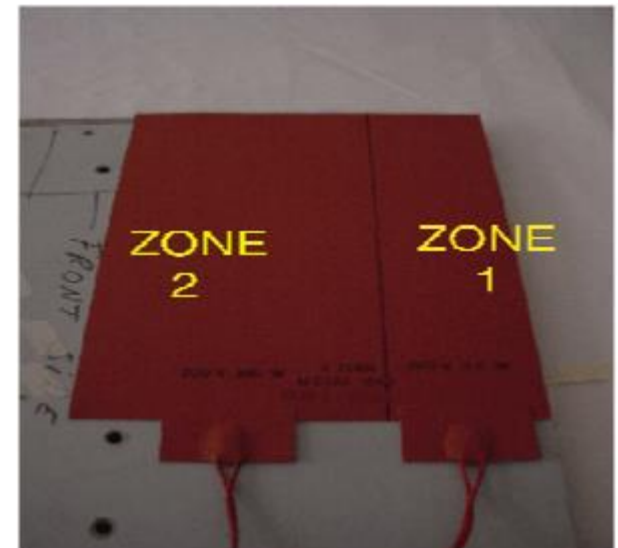
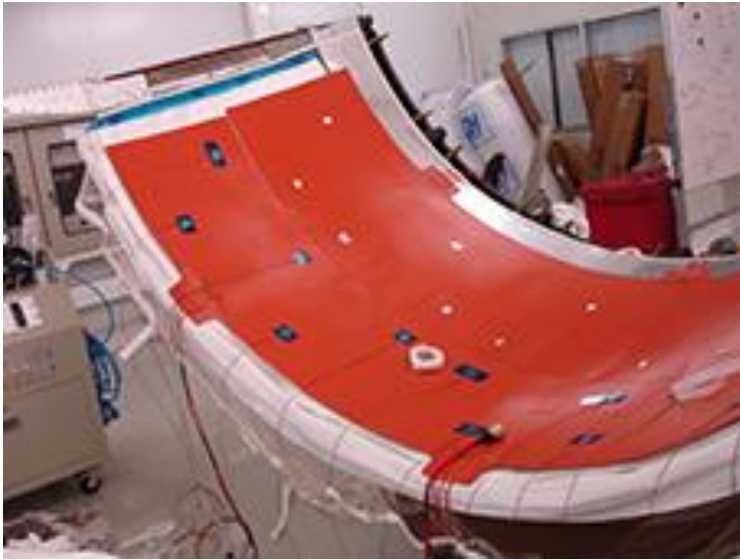
Structure below or near the repair may act as “heat sink”, by transferring heat to the environment. These areas are better identified during the “heating survey”, which should be performed **BEFORE** the actual repair. To compensate for these losses, additional heat should be offered to these areas, either by increasing the overlap length or by an additional blanket (see GMI’s blanket inventory). **A multi-zone heating strategy should be considered, as well.**

Multi-zone Heating Blankets



Example of multi-zone configuration, used for other demanding repair heating applications and corresponding heating blanket manufactured according to the definition calculated for this structure

Multi-zone / Shaped Blankets

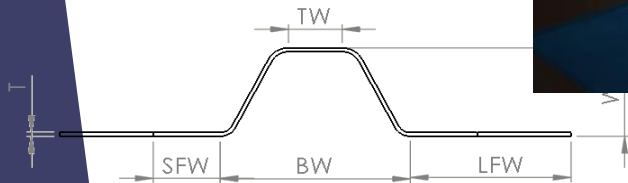




A350/B787 Increased complexity

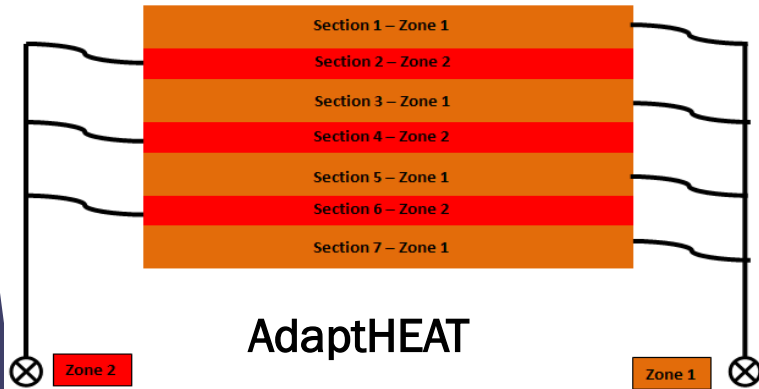
Thermal issues associated with A350 / B787 Panels repair

- Significant temperature fluctuation due to “Omega stringers” effect: trapped air acts as insulator
- Single heating zone approach is no longer valid.

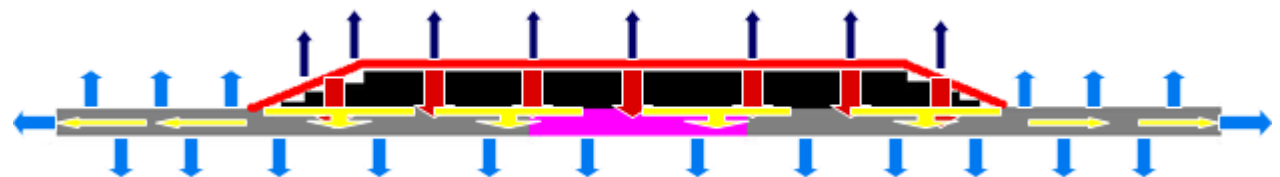


Zonal Approach is required...

Adaptive Heating



AdapTHERM solution
designed for A350
Omega stringers



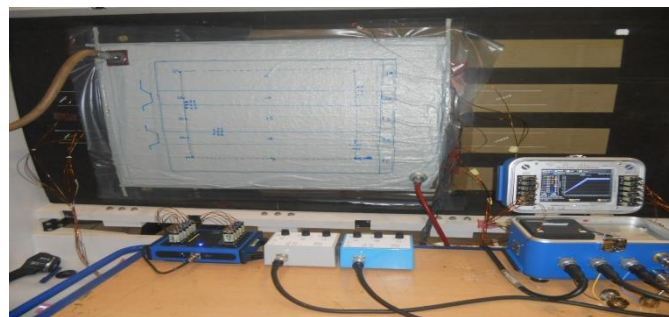
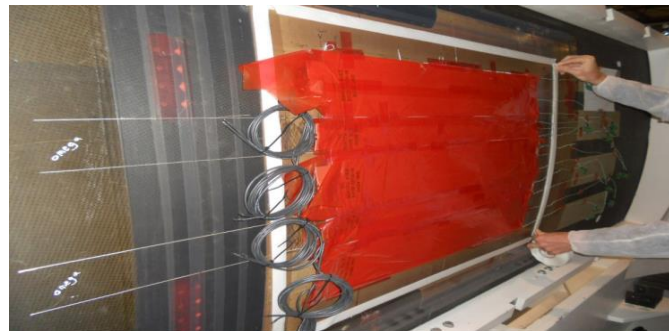
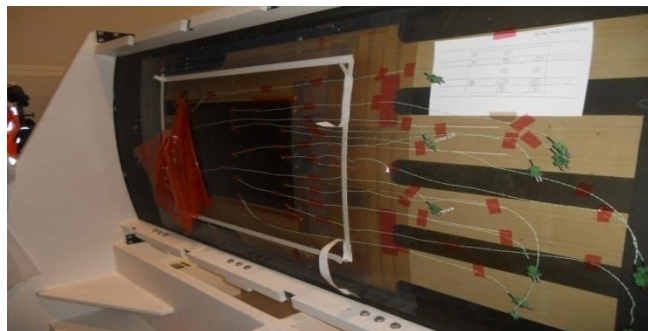
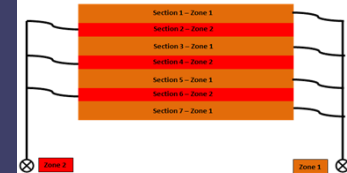
Heat Equilibrium = Algebraic Sum of Thermal
Routes No 1+2+3+4



GMI
AERO

Commercial in confidence

Adaptive Heating



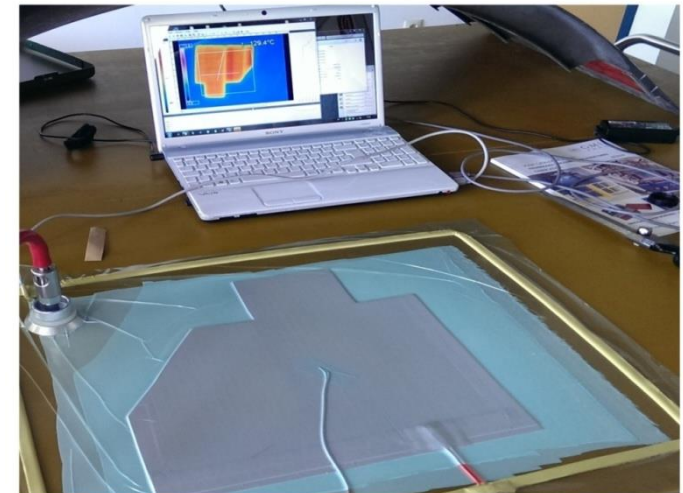
AdaptHEAT: Overall A350 panel application example

- Specially designed to ADVANCED / ANITA legacy equipment (no need for upgrade)
- Heating elements' geometry adapted to A350/B787 composite structures (fuselage / wings)
- Heating performance optimized for Ω /T/I stiffened composite structures
- Compatible with 2- zone controllers of all bonding console manufacturers (adaptors required)
- Standard procedure followed for preparation and vacuum bagging operations
- No need for additional training of operators
- Flexible dimensions, according to requirements
- Further customization possible, according to repair area configuration

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Blankets

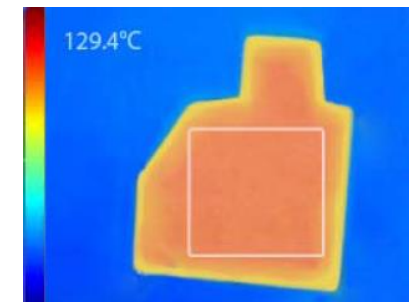
Heating Blanket “Cut-to-Measure” CONDUCTOR



Conductor Blanket cut to measures under verification test

Developed
and tested
under the
guidance of

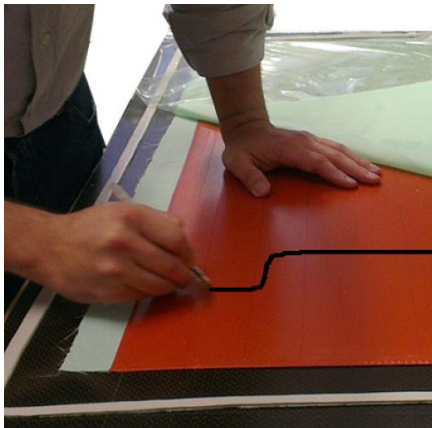
AIRBUS
GROUP
INNOVATIONS



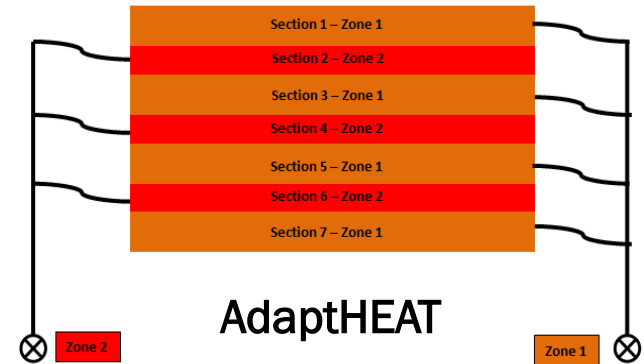


Blankets

Heating Blanket “Cut-to-Measure” **CONDUCTOR**



Combined with
AdaptHEAT for
A350 repairs...

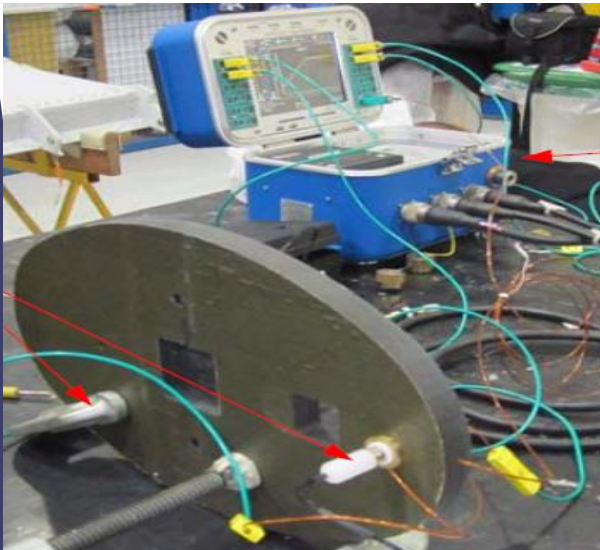
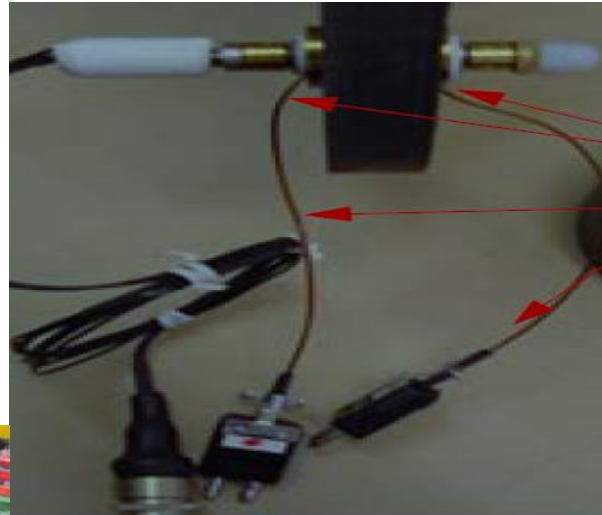




GMI
AERO

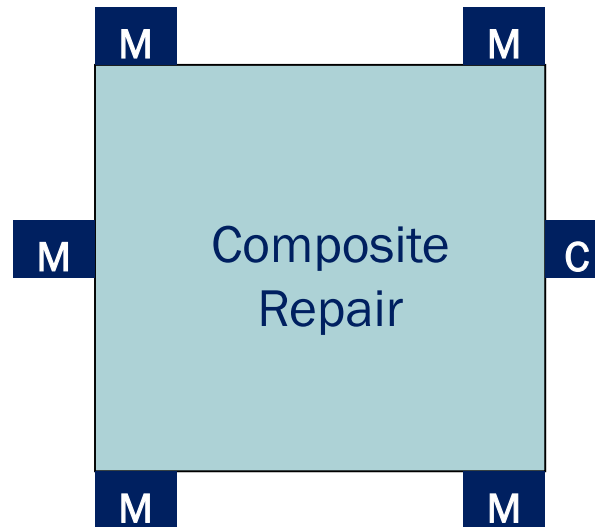
Special Heating Elements

Heating Pins - Heating Bolts for repair inside drilled holes
of 3-20mm diameter

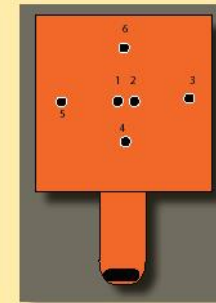


A GMI Innovation : The Sensing Mat

Install the control thermocouple very close to the repair



A Quick Connector is mounted on the mat to connect to the cable in one operation. Only the thermocouples involved in the process, selected by their right position above the patch, are connected to the bonding console.



Typical positioning of sensing points

The picture shows 6 sensing points organised on the surface of this mat. Each point is identified by its number. According to the repair patch dimension, the operator will decide which are the sensing points to select for the control and monitoring. These points will be connected to the console through its individual cables. The large mat will have 12 points embedded, while smaller will have 6 points.

To succeed good control you need to put the control thermocouple "C" at an area of "representative" temperature. As such, a point very close to the edge of the repair is usually proposed. Take care to put the monitoring thermocouples "M" around the repair and definitely at areas with potential heat sinks.

Ideally, use the GMI's "sensing mat", in order to be able to get temperature measurements from the center of your repair, without marking your part.

A set is constituted by a Mat and a Cable of 12 couples.

The P/N of the blanket must be selected according to the usual sizes of blankets as used by the operator.

The STSM dimensions must be larger than the blanket ones. Of course, one given STSM is suitable for all Heating Blankets of smaller sizes.

ANITA COMPATIBILITY

The STSM is manufactured to be fully compatible with ANITA and its thermocouples are thus of type J.

The STSM can be used with all types of ANITA : NG or OT, HE explosion proof models, 4, 6 or 8 zone Models.

The STSM is compatible with 1 or multi-zone GMI heating Blankets.



Select a Mat according to the dimensions and add a Cable. You can select one cable for more than one mat.

DIMENSIONS OF STSM MOST USUALS

You can order according to metric or imperial dimensions. All mats have 12 thermocouple type J embedded, but the smaller one 10 x 10 in (250 x 250 mm) which has 6.

Dimensions in cm	Approximates Dimensions inches	P/N	Number of Thermocouples
25 x 25	10 x 10	GMIHBSM025025	6
30 x 30	12 x 12	GMIHBSM030030	6
40 x 40	15 x 15	GMIHBSM040040	6
50 x 50	20 x 20	GMIHBSM050050	12
60 x 60	24 x 24	GMIHBSM060060	12
30 x 50	12 x 20	GMIHBSM030050	6

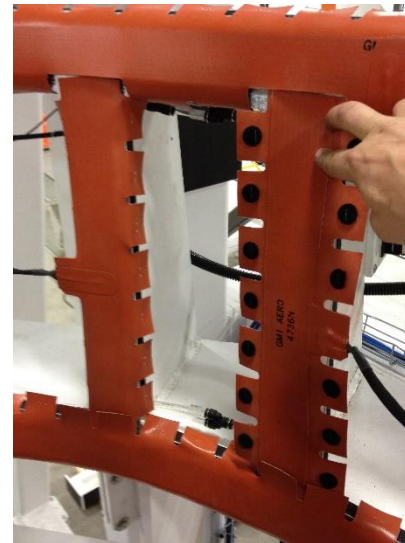
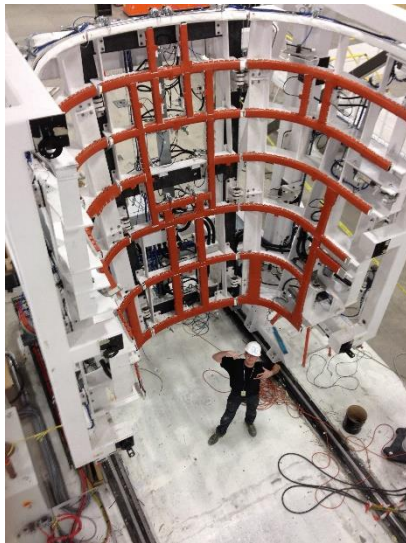
COMPENSATION CABLE

Making Cable : with capacity : 12 Thermocouples type J.
P/N : GMIHBSM-12J.

Making Cable : with capacity : 06 Thermocouples type J.
P/N : GMIHBSM-06J.

OOA

GMI has developed a solution to attach the complete metallic stringer set to reinforce the rear cone skin on Bombardier C- Series. These blankets are plugged on an Anita 6 zones.

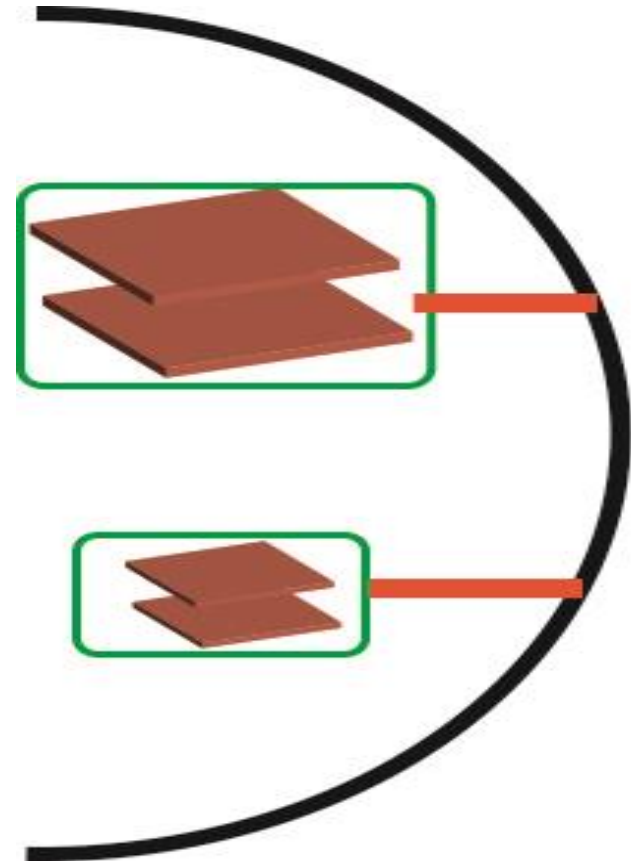




Repair Enhanced Autoclaves

Using flexible blankets as the main source of heating!

- Addressing repair requirements of complex aeronautical parts, such as the A380 engine reverses, the B777/B787 engine nacelles etc.
- Globally heating the parts in “conventional” autoclaves would **immediately distort** them while causing **severe and unrecoverable damage**, due to **variation of thermal expansion coefficients** of involved materials.
- Even though an autoclave is still used to provide required pressure and slightly elevated ambient temperature (i.e. 50-80°C), **heating at 120-180°C is ensured using specially designed adapted heating blankets, at the repair area ONLY!**

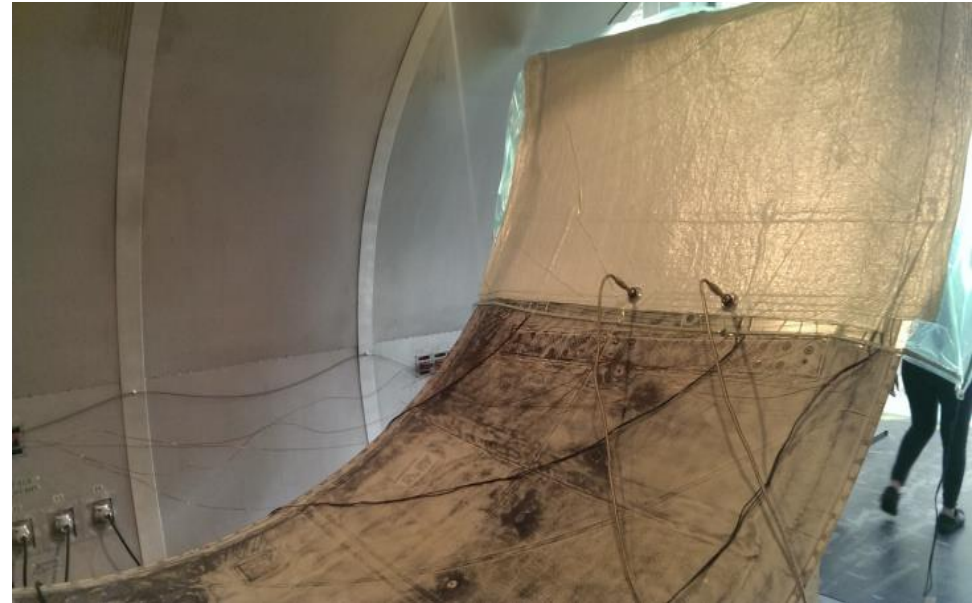




Repair Enhanced Autoclaves

Using flexible blankets as the main source of heating!

- Installation of power supply plugs, connectors, extension cables.
- Thermal **simulation** and **analysis** of the repair area to ensure Temperature homogeneity.
- Design and manufacturing of adapted **heating blankets**, tailored to the specific repair requirements.



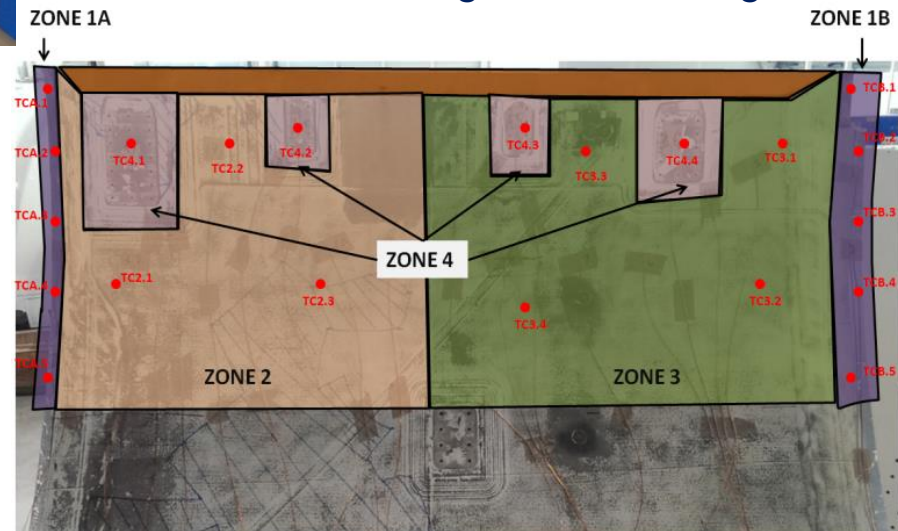
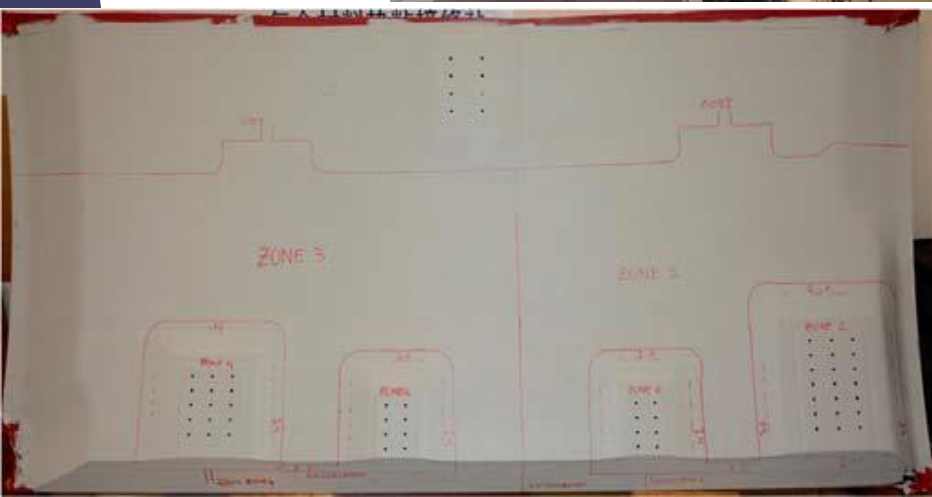
GMI – AFI cooperation for B777 GE90 Nacelle



Overview of the
GE90 area to be
repaired, showing
variations in
construction
materials and
methodology
(sandwich –
monolithic)

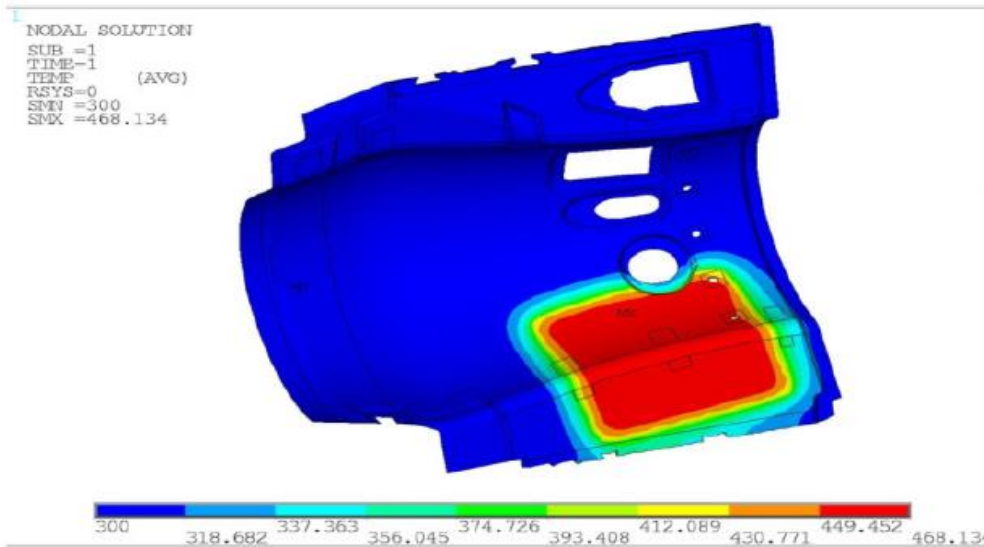


The 8-element / 5-zone adapted heating
blanket, designed to cover the upper
surface of the right / left cowling

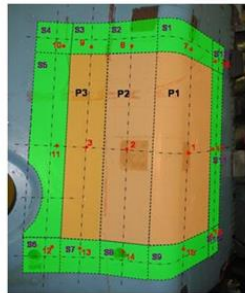
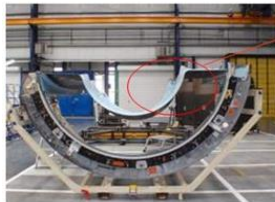




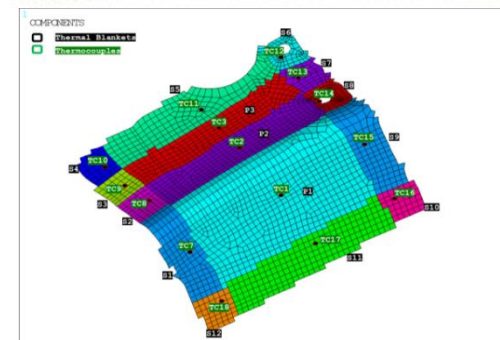
“ADVANCED” Out-Of-Autoclave (OOA) Heating Solution for Production and Repair of A380 Thrust Reversers using 18 zones



- Heating configuration defined (15 heating blankets on coured side + 3 on opposite side)



- Up to 72 T/C available for regulation and monitoring
- Some of them could be localized specifically for correlation with Thermal Simulation



Repair Enhanced Autoclaves



“ADVANCED” Out-Of-Autoclave (OOA) Heating Solution for Production and Repair of A380 Thrust Reversers

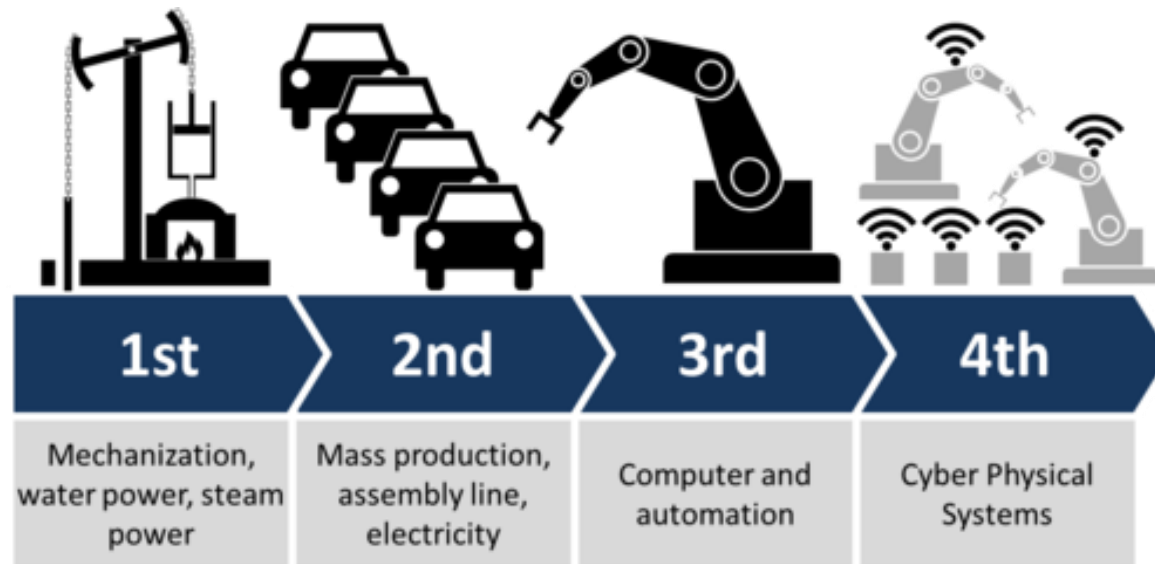
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Answer to Challenges:

- a. Adaptive Heating Solutions
- b. Integration in 4.0 concept**
- c. Training & Continuous Professional Development

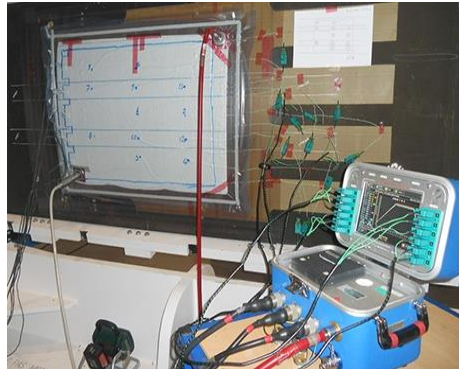
ANITA 4.0 is the application of the
Industry 4.0 concept in the field of
bonded composite repairs



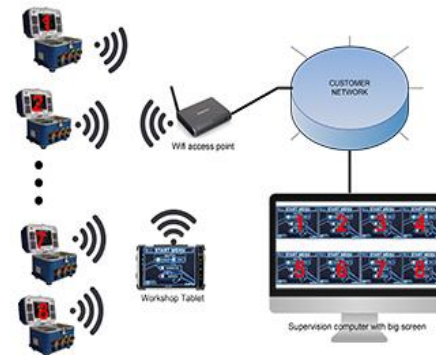
ANITA 4.0 concept Main Steps



Last steps of
repair preparation
on an A350
fuselage



ANITA EZ with
AdaptHEAT ®
controlling the
curing process



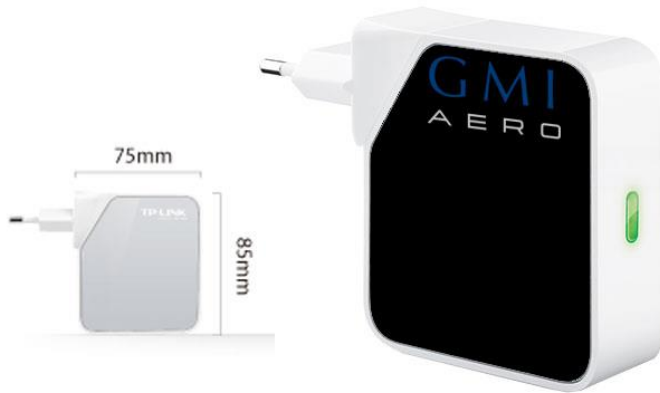
ANITA EZ
connections to the
company's
intranet



On-line inspecting
and evaluating the
overall process

Proximity WiFi Solution

Remote Monitoring and Control Tablet



Technicians could be engaged with other activities in their offices or around! → 80% reduction of MH associated with repair...



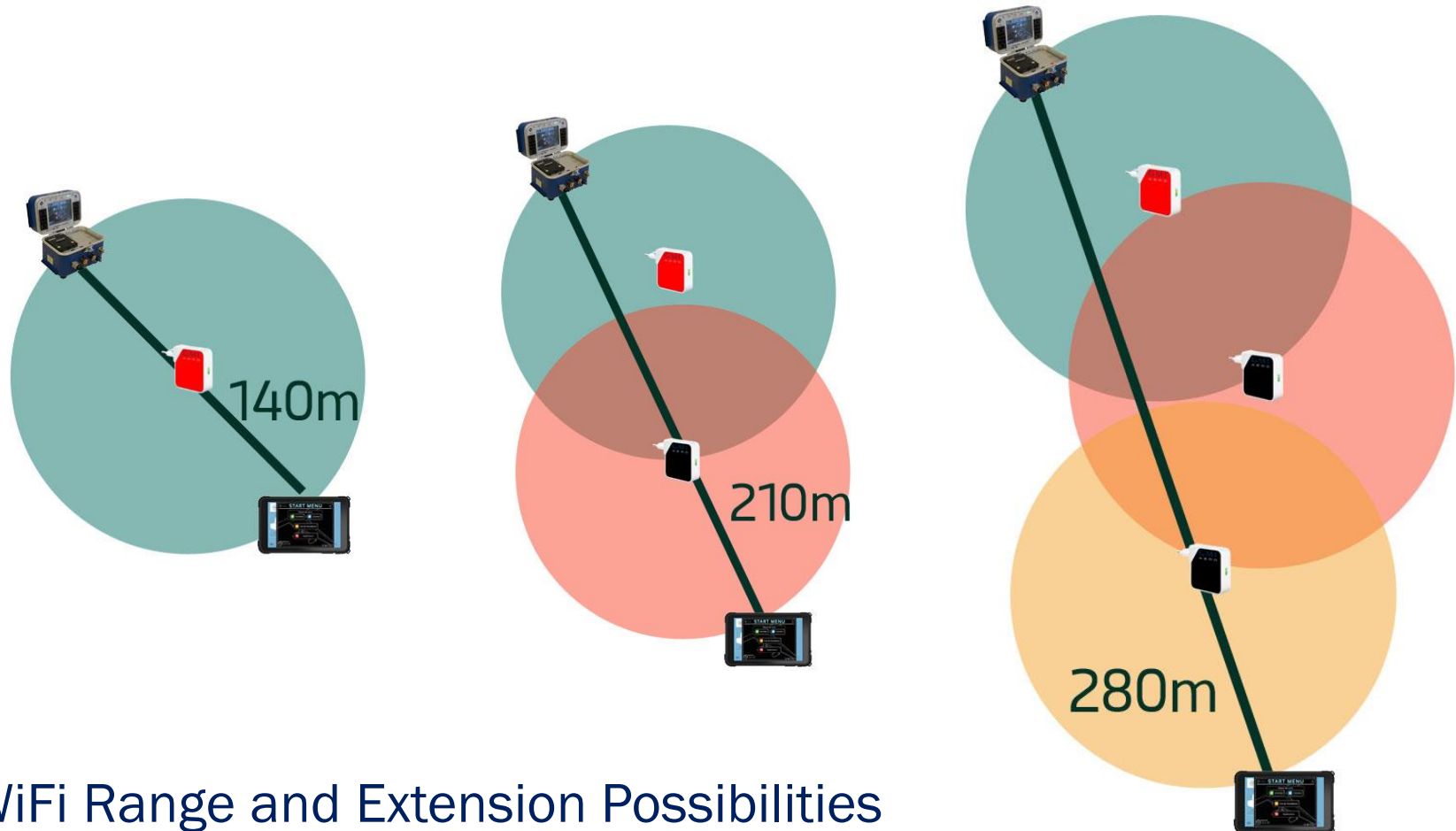
Remote Control Tablet



Surveillance and control of up to four (4) bonding consoles

Proximity WiFi Solution

Remote Monitoring and Control Tablet

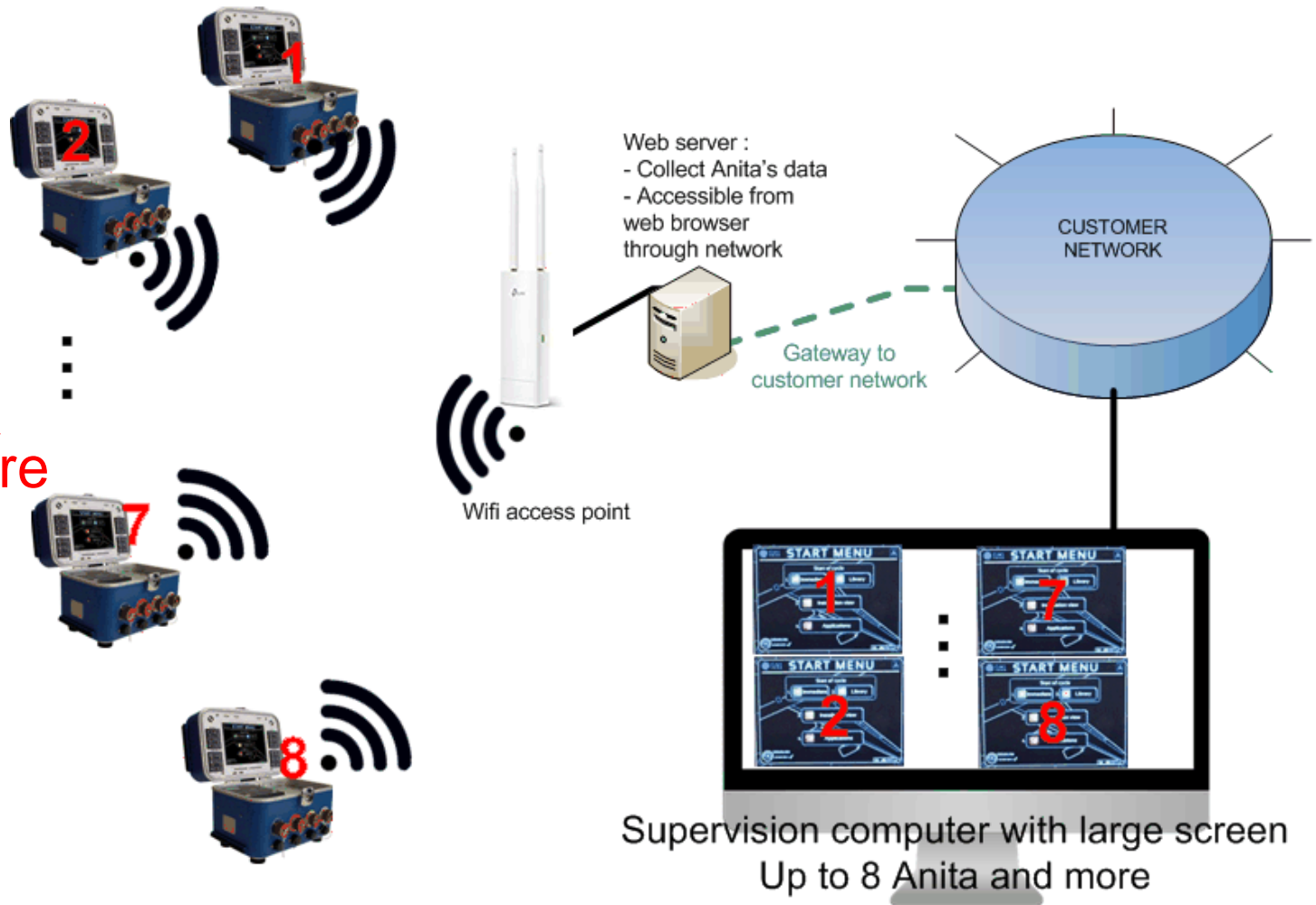


WiFi Range and Extension Possibilities

Distant IT Solution

PANOPTES

General Network Architecture

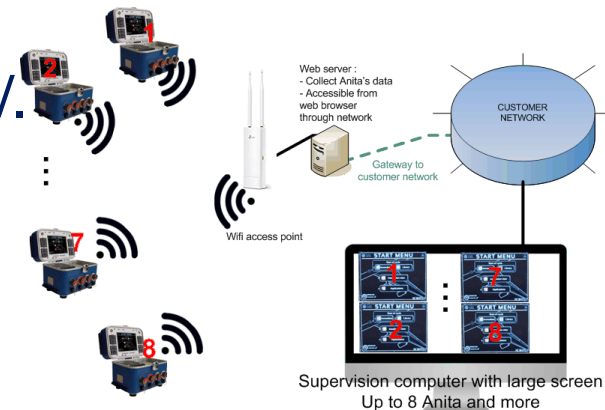


Distant IT Solution

PANOPTES

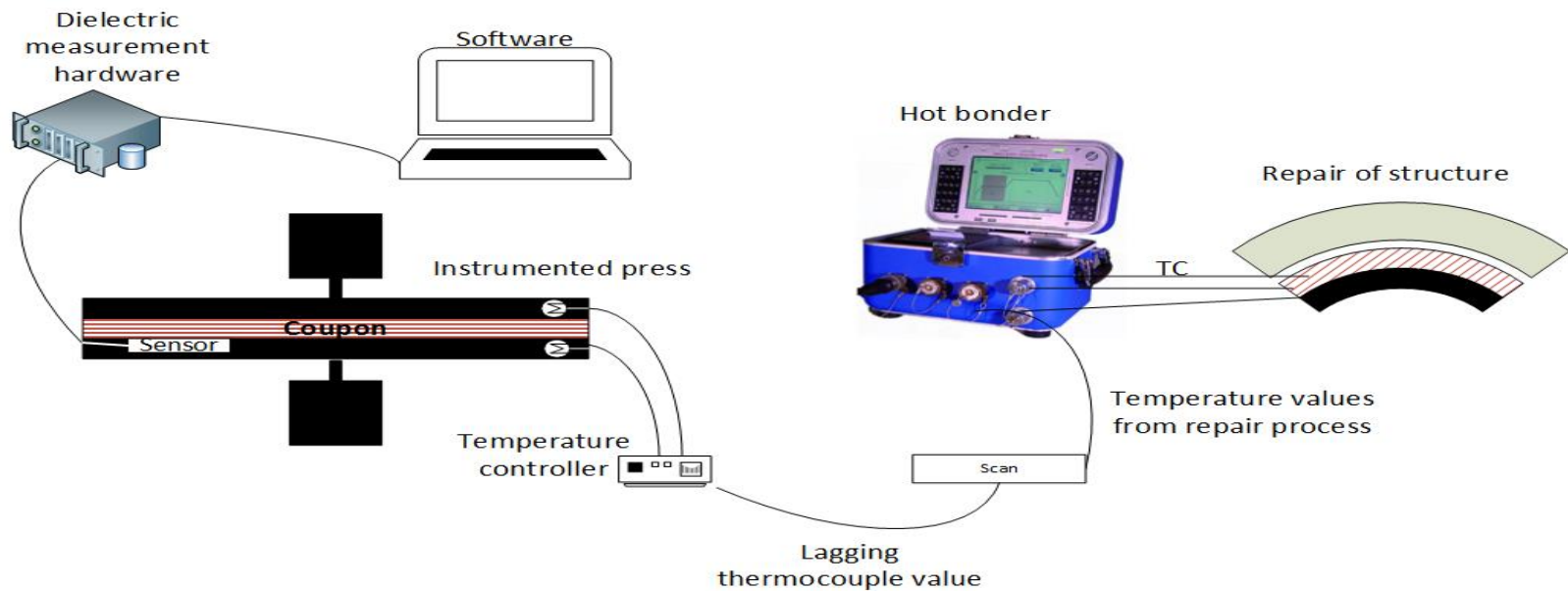
Distant Parallel Monitoring

- Up to **eight (8) ANITAs** connected to end-user's central IT system through WiFi.
- **Web application**, accessible from main web browser.
- Collects ANITAs information & stores in a **database**.
- **Visualizes** summary of all connected ANITAs current status (Zones on cycle, Setpoint, T/C, Vacuum, etc).
- **Data presented** on PC screen or Smart TV.
- Format and variety of presented data **tailorable** to end-user's requirements.



Repair Digital / Physical Twin

On-line Quality Monitoring



- When a **safety critical** structural repair is performed all repair data (temperatures, vacuum level etc.) are **recorded** and **real-time transmitted** to the aircraft manufacturer.
- At the aircraft manufacturer's facilities a **setup using same materials** is prepared and is **simultaneously** to the actual repair **cured**, using **appropriately selected portion of transmitted real time data** (e.g. lagging thermocouple), so as to enable **imminent destructive or non-destructive testing** of produced material.
- Within the set-up prepared at the aircraft manufacturer's facilities, **dielectric sensors for curing degree monitoring** are included, thus providing **real time degree of cure data**, equally valid for the remotely performed structural repair on the aircraft.

Answer to Challenges:

- a. Adaptive Heating Solutions
- b. Integration in 4.0 concept
- c. Training & Continuous
Professional Development**

Aircraft Composite Repair Evolution and Challenges

BUTCHER OR BAKER?

There are some significant challenges for technicians due to the nature of composites - which are very process driven and rely heavily on the composite technician building quality into the finished product. To illustrate this, the Aug/Sep 2012 edition of Aviation Maintenance (AVM-mag) published an article discussing the difference between repairing a metal vs a composite aircraft being akin to the difference between highly skilled butchers and talented bakers. Here's why:



Image source: dreamstime.com

Butchers don't have to create the meat they work with. The raw meat comes into the shop fully formed. The butcher's job is to carve it up into the required cuts. This is similar to mechanics working on a metal aircraft. Whether they are installing a new part, re-shaping and repairing an old one or - in extreme cases - custom measuring, cutting, shaping and custom fitting a new metal part from scratch, mechanics don't have to smelt and cast the metal they are working with. Instead, the metal comes into the shop in workable shapes, sheets and thicknesses, no fundamental manufacturing is required. It's not quite as convenient as working with meat, but it is close.

In contrast, bakers must start with the raw ingredients, condition them to room temperature, and combine them into specific mixtures first. Then they have to transform these mixtures into cakes, pies, breads, etc. and use the right levels of heat for the correct amounts of time. In the same vein, composite repair technicians must follow strict processes to combine and cure ingredients to achieve an optimum result.



GMI Training Activities



- **Level 1: BASIC BONDED COMPOSITE REPAIR STANDARD TRAINING COURSE – 5 DAYS**
- **Level 2: ADVANCED BONDED COMPOSITE REPAIR INTENSIVE TRAINING COURSE – 5 DAYS**
- **Level 2: ADVANCED BONDED COMPOSITE REPAIR STANDARD TRAINING COURSE – 10 DAYS**

Example of 5 days Basic Training Course

Level I – Basic Bonded Composite Repair

Scope: This curriculum is intended to meet the formal training requirement for individuals who intend to become certified as aircraft composite repair technicians. Persons who successfully complete this aircraft structural repair training program are considered to be able to perform **basic** composite bonded repairs to aircraft structures in compliance with the manufacturers' repair documentation or other acceptable or approved repair data.

Applicable documents: GMI Aero proposed Training Courses align with the latest SAE Commercial Aircraft Composite Repair Committee (CACRC) and regulatory authorities documents, as follows:

- **AIR4938C** Composite and Bonded Structure Technician/Specialist Training Document
- **AIR4844D** Composites and Metal Bonding Glossary
- **AIR5719B** Teaching Points for an Awareness Class on "Critical Issues in Composite Maintenance and Repair"
- **FAA AC 65-33** Development of Training/Qualification Programs for Composite Maintenance Technicians

Trainee's prerequisites:

- Experience as aircraft maintenance technician or engineer
- Good knowledge of the English language (minimum B2)
- Basic mathematics, physics and chemistry considerations (minimum High School - level or equivalent)

Course Outline: This training course is divided into four (4) main training units, namely:

- Basic Theoretical Considerations
- Bonded Composite Repair Equipment and Toolings for Curing and Machining
- Implementation of actual Bonded Composite Repairs
- Theoretical & Practical Examinations



EASN &
GMI Aero
CPD
Seminar on

**Bonded Composite Repair of Aircraft Structures:
Contemporary Challenges and Latest Innovations**

When: 18-20.06.2019

Where: Paris, France

Programme Content

The recent venue of all - composite fuselage aircraft (A350 - B787), together with the expansion of older aircraft fleets, introduces new requirements in bonded composite repair. These contemporary repair challenges and the latest innovations in equipment and methodologies to address them, will be the subject of a **2-Day CPD Seminar**, organized by EASN Association in cooperation with GMI Aero, the lead European composite repair equipment manufacturer.

Key Benefits

Experienced industrial personnel, together with academic experts, will provide an insight of recent R&D developments and critical issues in the field of bonded composite repair, ensuring that seminar participants acquire a full "process understanding", including a **demonstration workshop**, to support reliable application of bonded repairs, even on Class I (safely critical) structures.

Key Note Speakers

Two key-note speeches will be presented by experienced engineers from major aircraft manufacturers and MRDs.

The participants will receive a **Certificate of Attendance** highlighted by the added value of the EASN Association quality label.



Practical Information

Technical Visit

Participants will have the opportunity to join an optional visit to Le Bourget Air Show on a Date reserved for Professionals (20.06)



<https://www.siae.fr>

Fees

- EASN members € 500,00
- Non-EASN members € 600,00
- Students € 350,00

Amounts are VAT excl. (20% VAT will be added, where applicable)

Contact & Registration Details

For further information on CPD seminars please visit our website:
www.easn-tlc.com/cpd



Continuous Professional Development



CPD Participants, Organizers and Speakers in action, during the Seminar

**“New Aircraft, New Materials,
New Repairs”**

Guillaume FERRER

*Embodiment Industrialisation Manager
Composite Repair Process Development
Airbus Customer Services*



**“Technical challenges and innovations in
bonded aircraft composite repair”**

Philippe SERVANT

*Responsible for Engineering, Research
and Development of Aerostructures,
AFI KLM E&M*



[CPD Video](#)



SINGAPORE AIRSHOW 2020
ASIA'S LARGEST AEROSPACE AND DEFENCE EVENT

11-16 February 2020
Changi Exhibition Centre



GMI
AERO

**Composite Technology Seminar
& Experts Forum**
Singapore, 10/2/2020



GMI
AERO

**Composite Repair Seminar
& Experts Forum**
Berlin, 11-13/5/2020



Bonded Composite Repair Methodologies – Latest Advancements and Developments



Questions?

