

FABHELI

E-Lass Seminar Day
PORNICHET - 26/06/2018

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Office MECA

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Modelling department
SIREHNA – Naval Group

FABHELI

THE FILM ([link](#))

ACTORS

PARTNERS

LOIRETECH (Leader)

MECA

NAVAL GROUP

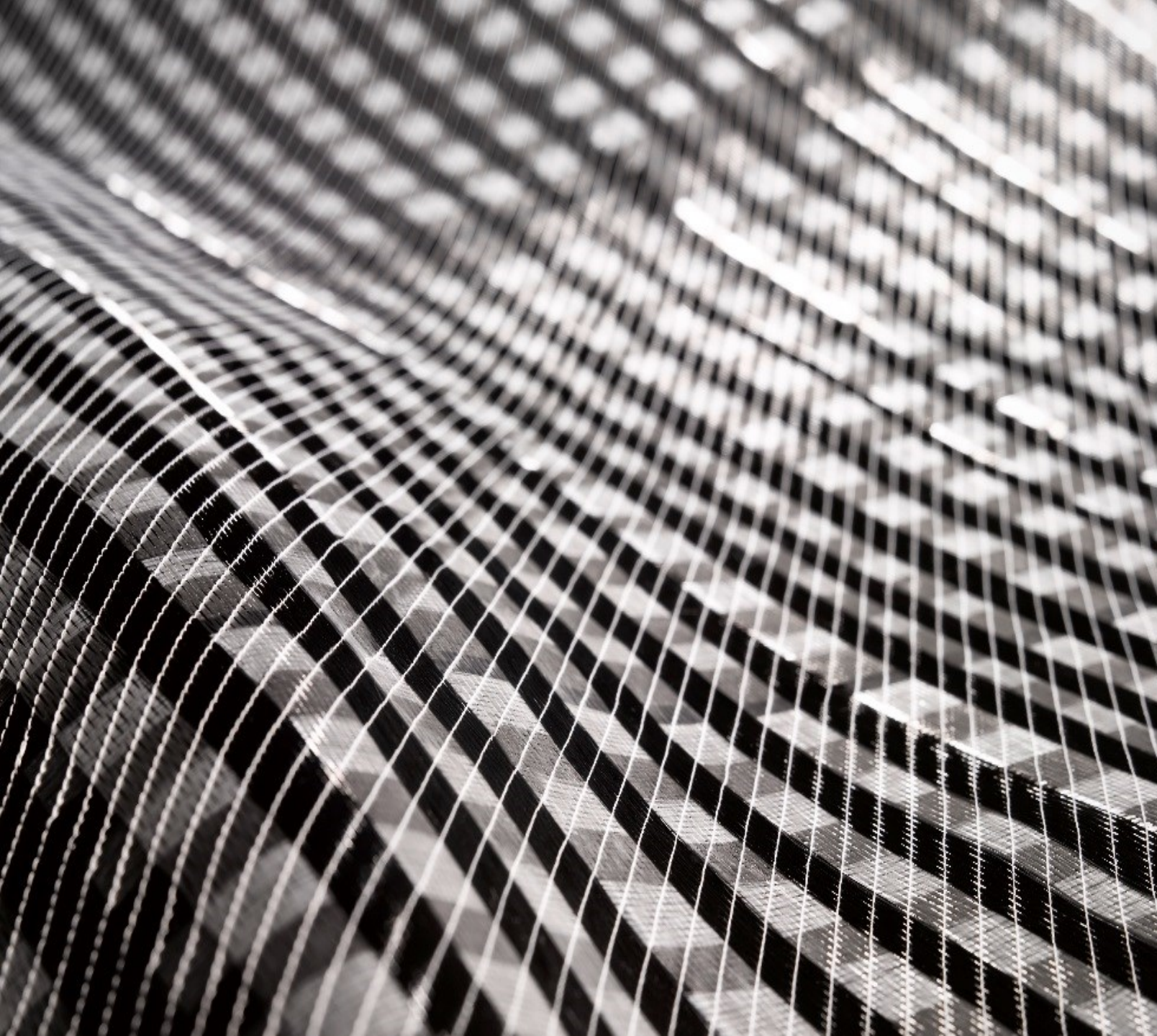
CONTRIBUTORS

Bureau Veritas

AML



mēca



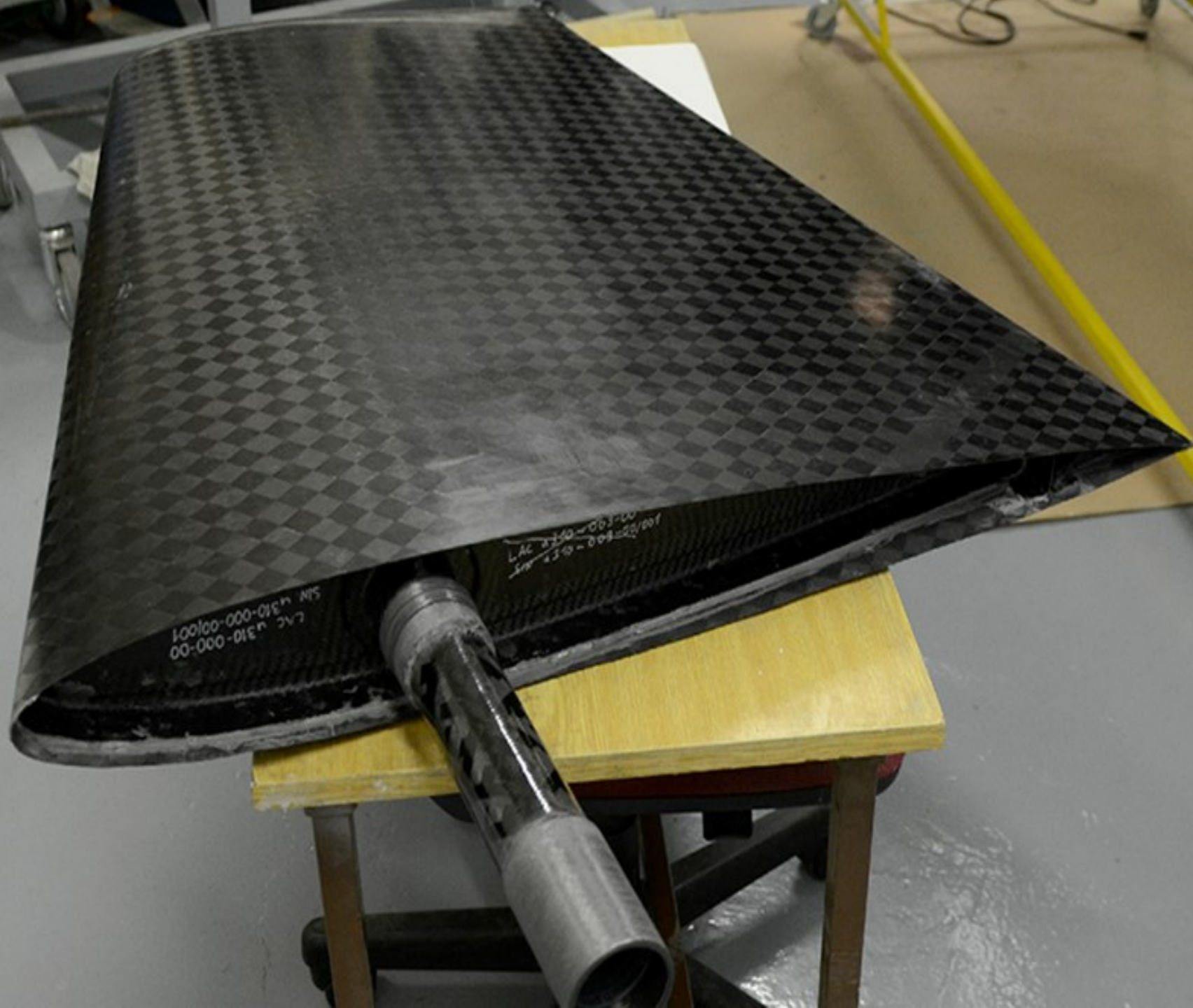
COMPOSITE MATERIALS ?

mélange de fibres ou renforts
et de résine

polymérisation de la résine

densité faible

procédé de fabrication
utilisant un outillage ou moule



GENESIS

Composite propellers or composite blades are classically manufactured using pre-preg technology.

- High cost,
- Weakness of the junction between profiles

THE PROJECT

Collaborative Research
Program - DGA Rapid

Objectives :

To design,

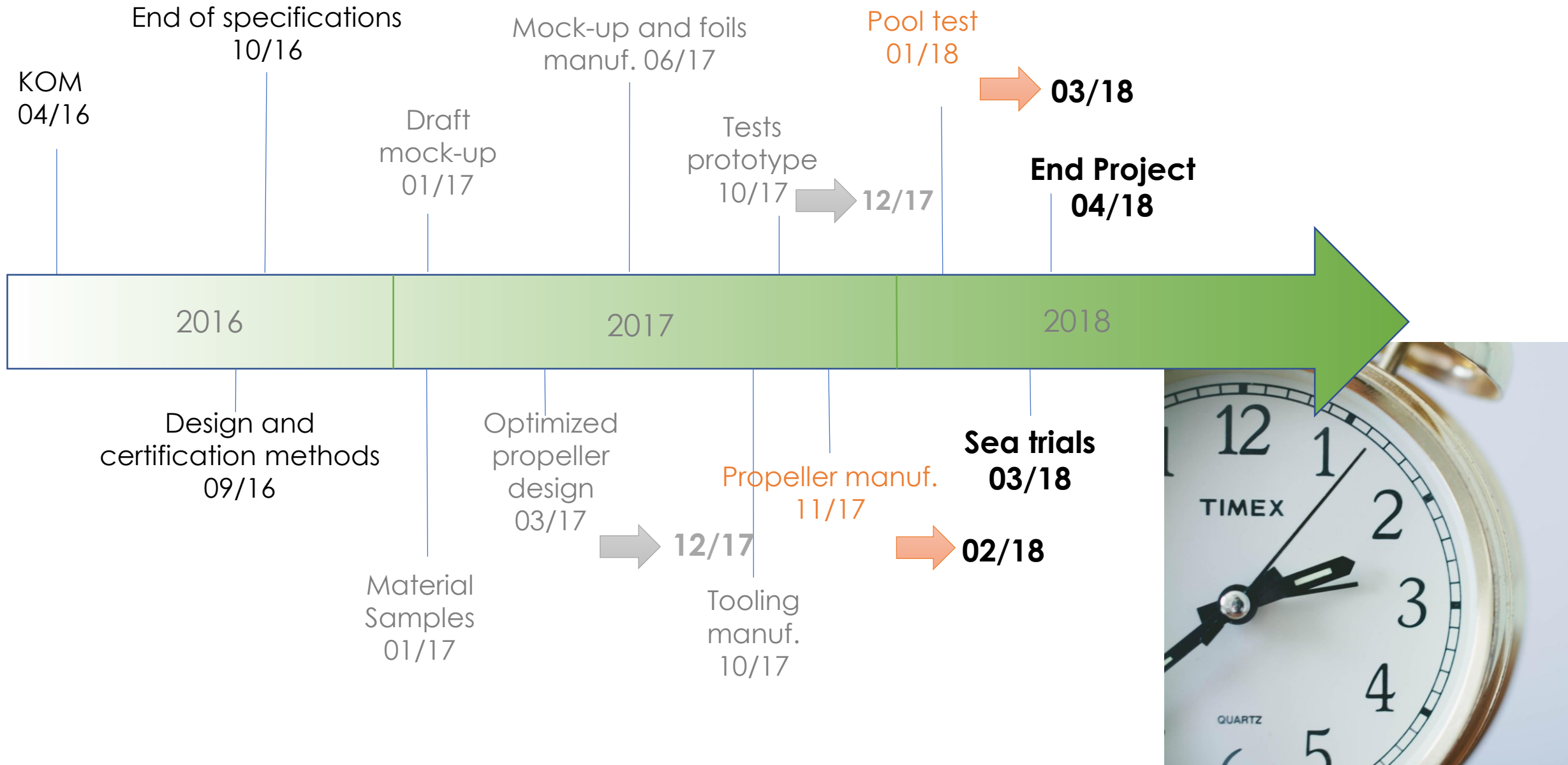
To manufacture in an
industrial way,

And to test in real conditions,

A propeller made of
composite materials.



A SHORT TIME



STRONG OBJECTIVES

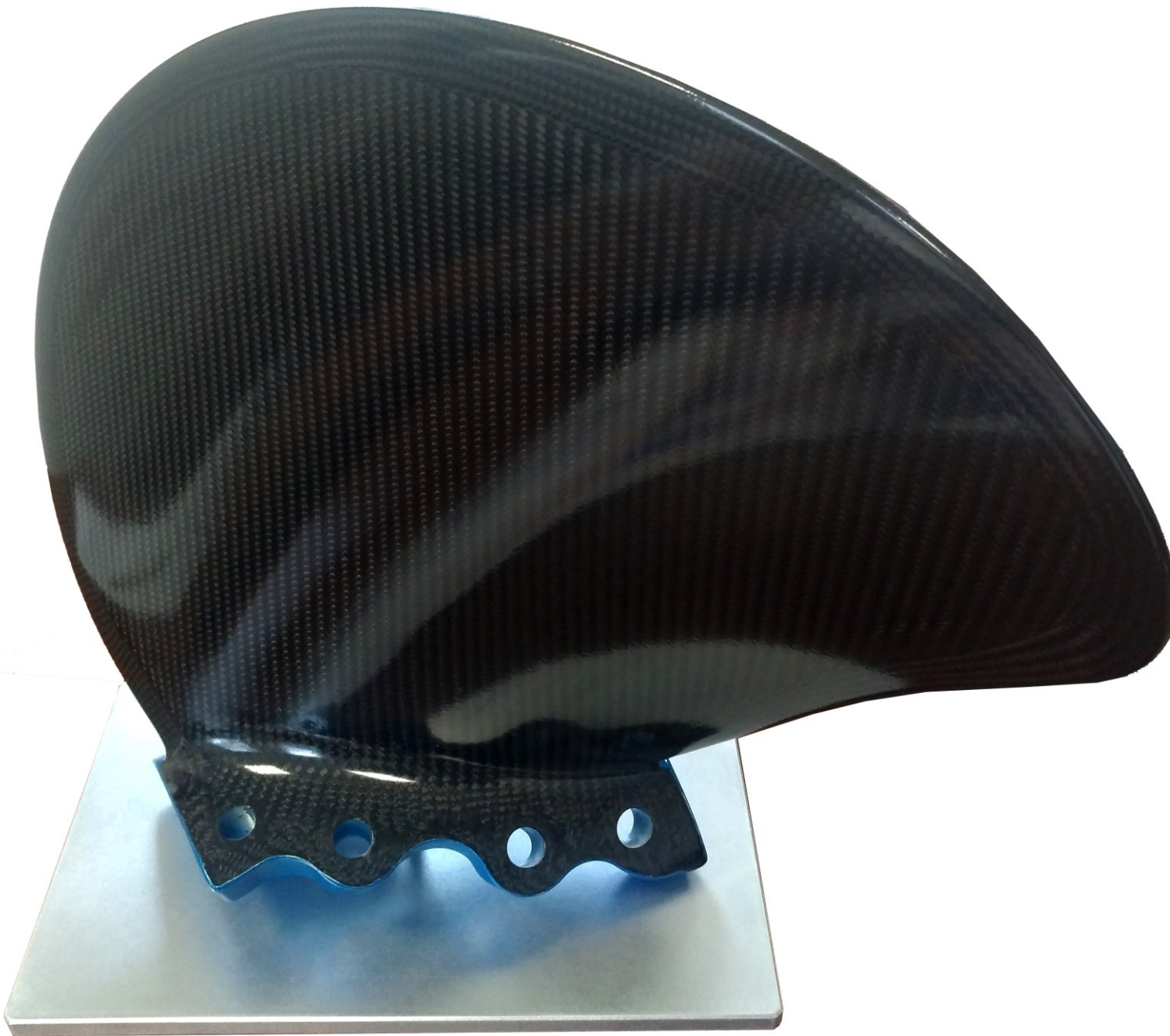
To replace a metallical propeller on a twin-propeller vessel

Diameter 1.1m

the PALAIS, a passenger ship 30 meters long and 84 tons with two engines of 1100 hp

To design removable propellers





LOCKS

Main

Development of the RTM manufacturing process,

Make "one shot" blades, without finishing operations,

Control of composite propeller costs,

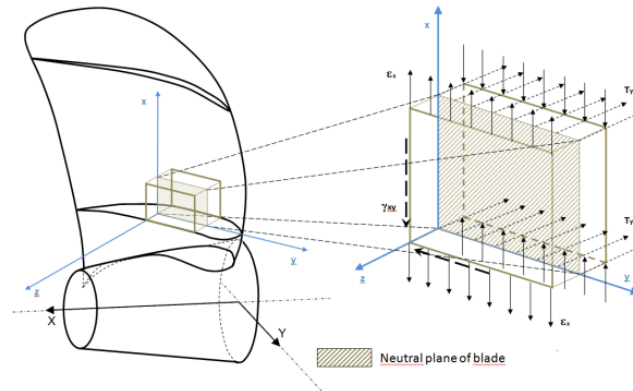
Robust blade / hub connection,

Impact performance at the leading edge,

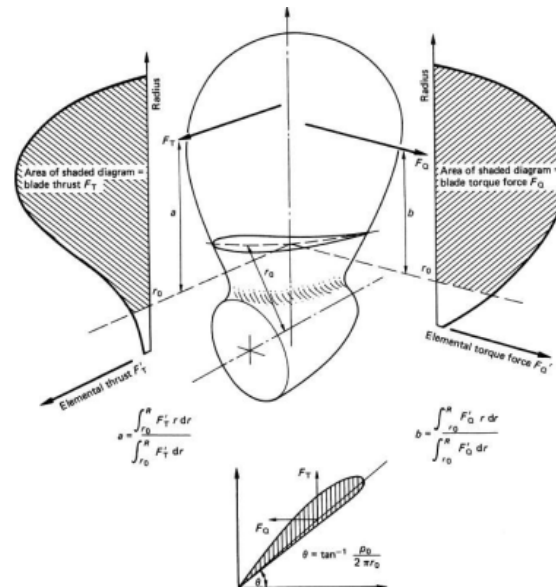
Propeller performance,

Normative framework or rules to build,

RULES OR STANDARDS



The partners called on Bureau Veritas, which oversaw the design of the propeller, preparing for future certification.



Fundamental work to go back to the origin of the dimensioning of the metal propellers, then to apply it to the composite propeller.

FabHeli - Composites Propeller

Lot 4 - Design methodology

Expertise Department / B. Collier-S. Paboeuf

DS/SEER/15/AT16061- 23/10/2017

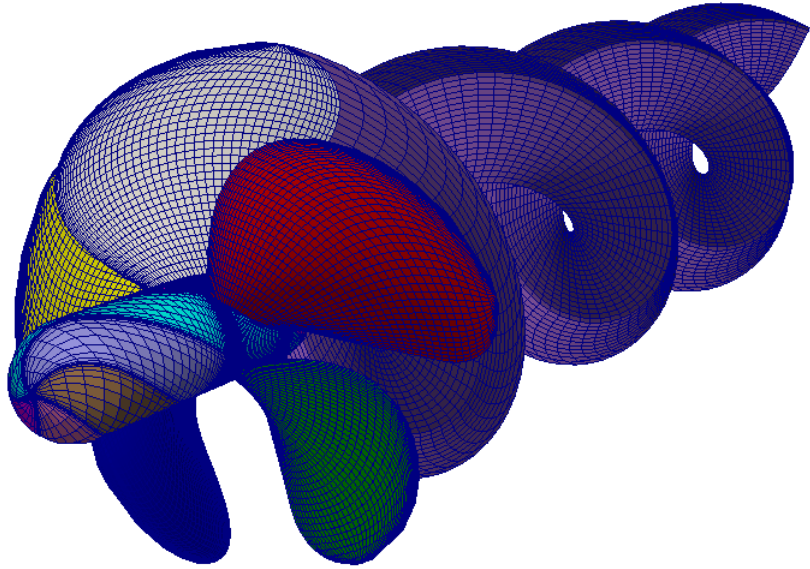


**BUREAU
VERITAS**

Move Forward with Confidence

FLUID-STRUCTURE COUPLING

Development of an automated fluid structure coupling



Flow solver PROCAL 2.309

3D Boundary Element Method (BEM) based on potential flow theory

blades, hub and blade wakes are meshed using quadrilateral panels

Structure solver ABAQUS 6.13 distributed by "Dassault Systèmes".

STRUCTURE SOLVER

Type of elements

Volume elements	Thick shells	Shells
<ul style="list-style-type: none">😊 High order elements😊 Accurate representation of the geometry (structure and shape)	<ul style="list-style-type: none">😊 Low computer resources needed😊 Accurate representation of the geometry	<ul style="list-style-type: none">😊 Low computer resources needed😊 Accurate description of the ply stack (specific to ABAQUS)
<ul style="list-style-type: none">😞 Elements must have the same thickness as the plies \Rightarrow too small	<ul style="list-style-type: none">😞 Adapt the description of the ply stack for each element	<ul style="list-style-type: none">😞 Loss of accuracy of the outer shape

Meshing and computing issues

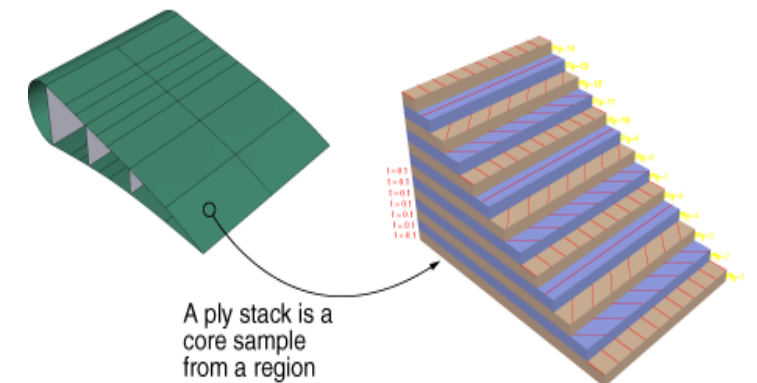
Ply description issue

Pressure and displacement transfer issues

These issues can be tackled with variable difficulty level

The ply description issue can be solved and automated easily

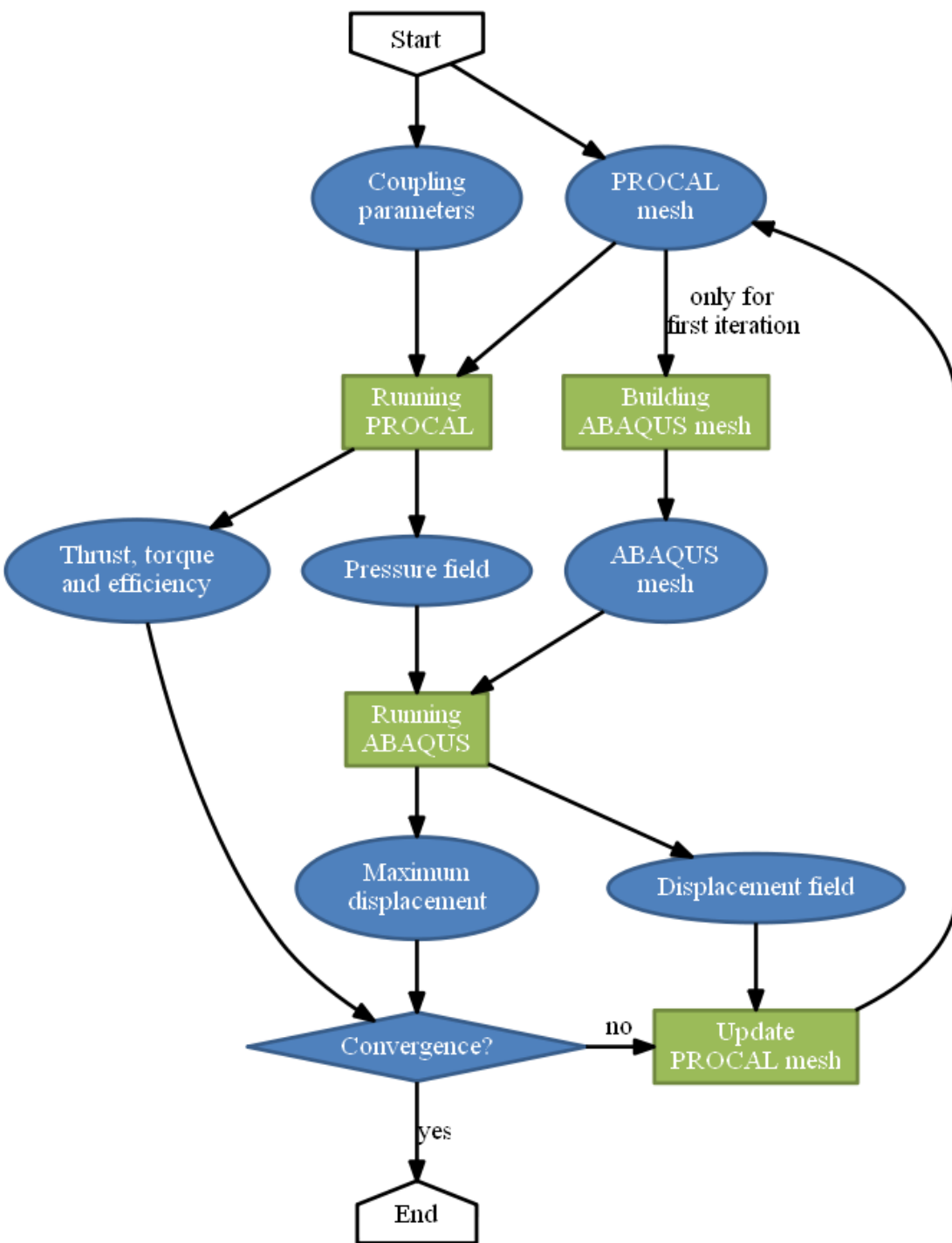
Thick shell elements have been chosen



COUPLING ALGORITHM

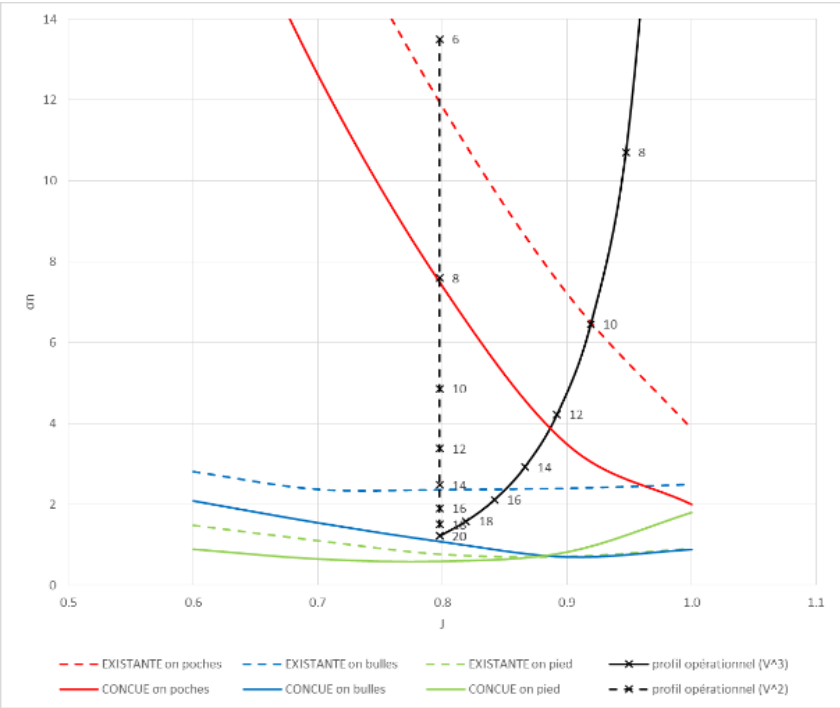
Based on iterations overs pressure and displacements until convergence

To prevent interpolations: the surface mesh is the same for the structure and flow solver



Résultats à 20 nœuds

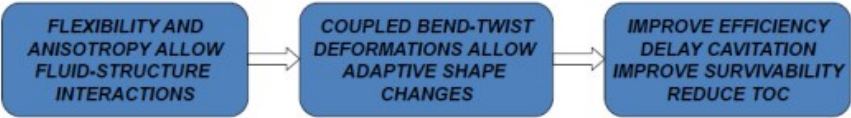
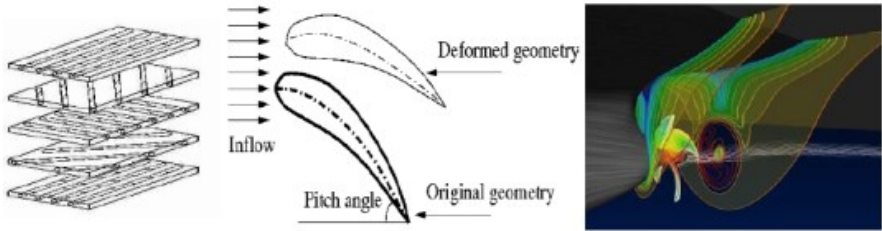
Grandeur	Unité	Hélice existante	Hélice conçue	Variation
Vitesse de rotation	RPM	731.1	730.1	-0.1%
Puissance au frein	kW	969.0	934.1	-3.6%
Puissance délivrée	kW	741.3	714.6	-3.6%
Couple	kN.m	9.7	9.4	-3.4%
10K _Q	-	0.4037	0.3912	-3.1%
J ₀	-	0.796	0.798	+0.1%
K _T	-	0.2054	0.2049	-0.3%
Poussée	kN	45.0	45.0	-
J _s	-	0.771	0.772	+0.1%
Rendement	-	0.645	0.665	+3.1%



PERFORMANCE

À 20 nœuds, l'hélice conçue est plus performante que l'hélice existante sur le plan du rendement et de la cavitation (réduction du risque de cavitation).

L'hélice composite est plus flexible, ce qui peut être mis à profit pour améliorer son rendement (variation de pas avec la déformation).





LEADING EDGES

Reinforcement of the leading edges to improve the resistance to impact without complicating the manufacturing process.

COATING

Application of a coating

Improved hydrodynamic performance

Improvement of resistance to cavitation

BLADE/HUB JUNCTION

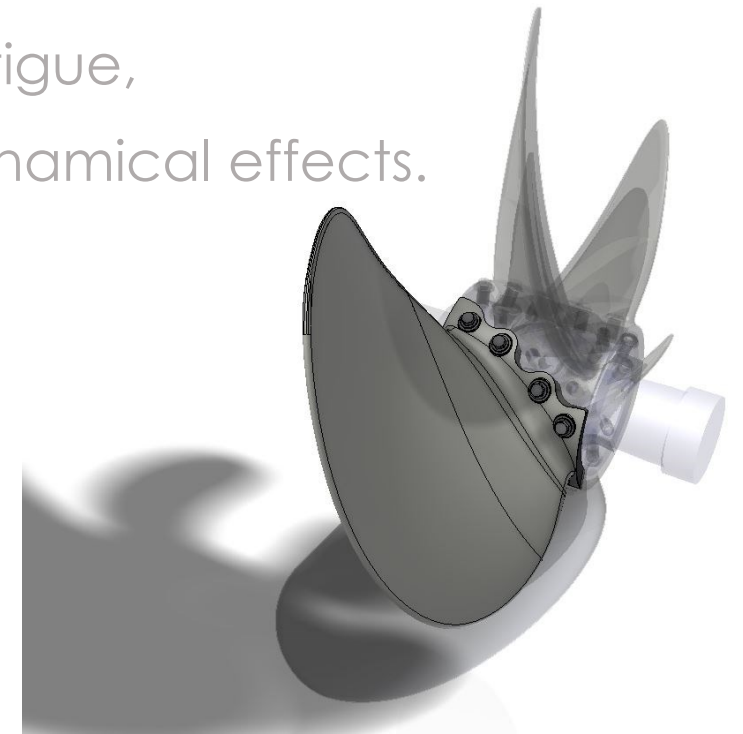
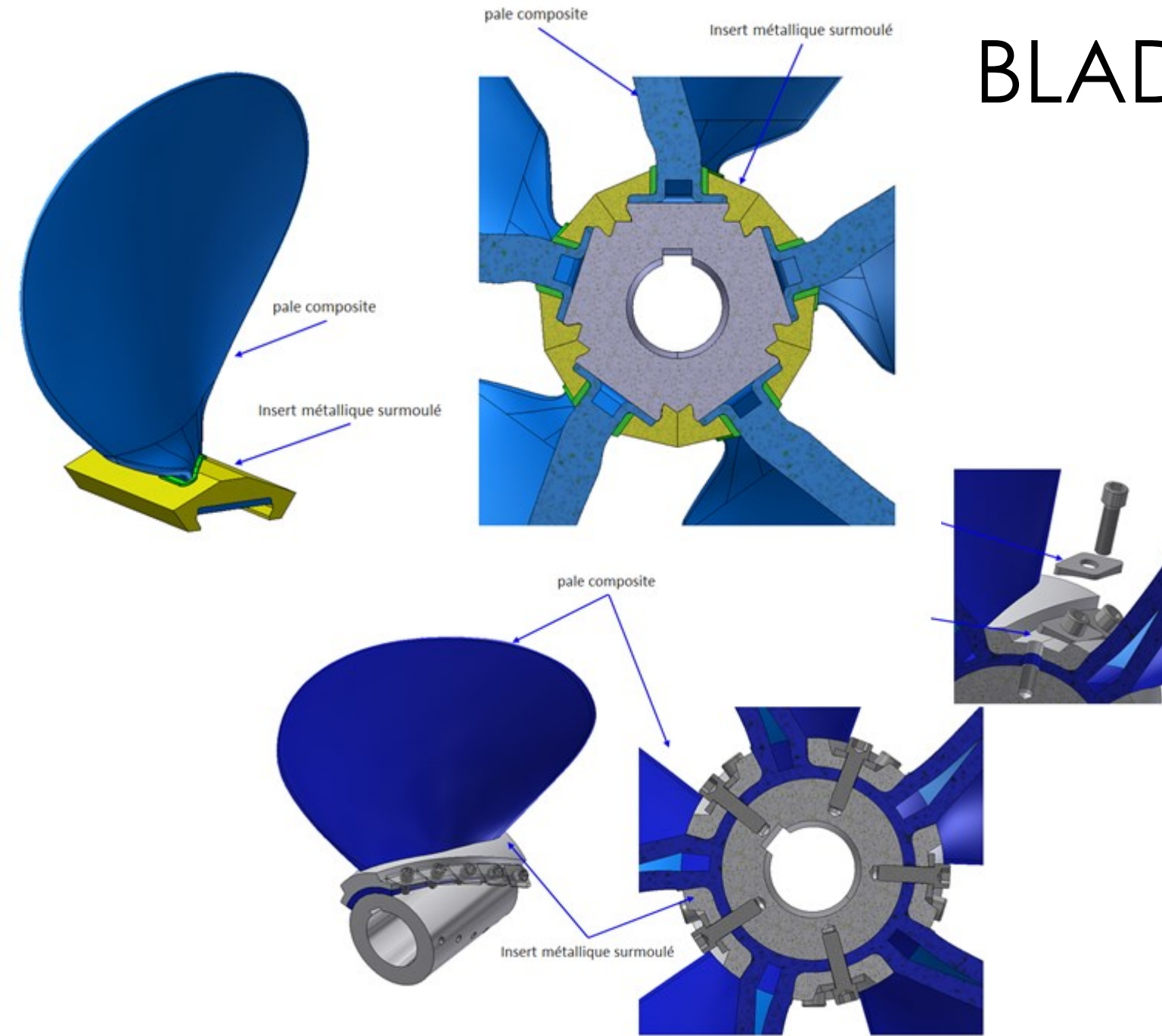
Search of mechanically reliable solutions, while ensuring disassembly.

Max torque motor,

Clamping,

Fatigue,

Dynamical effects.

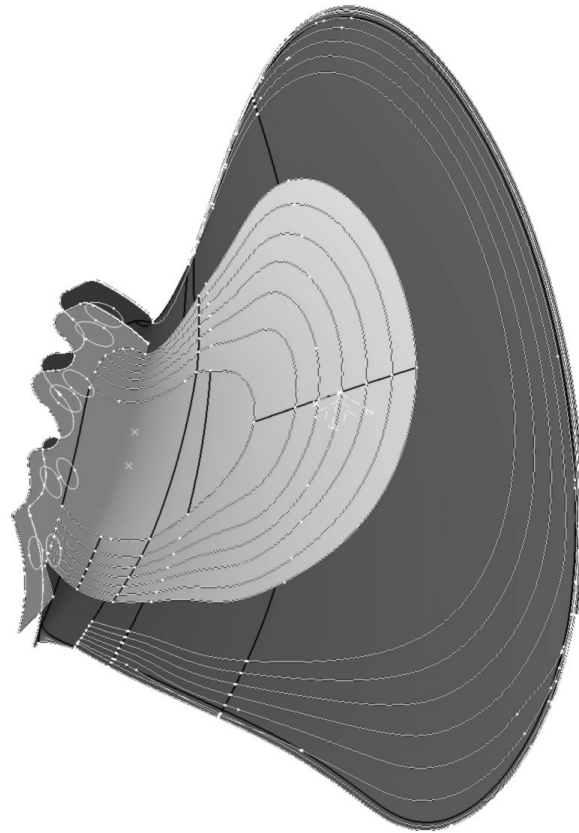
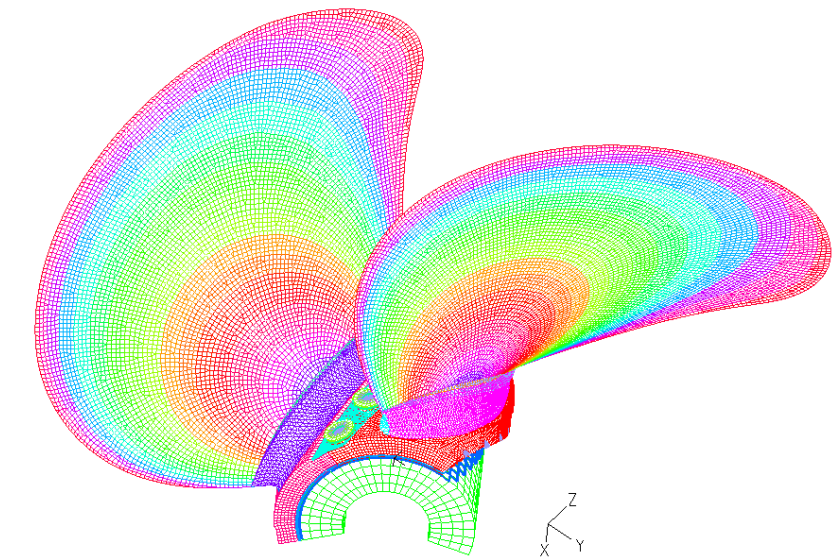


COSTS UNDER CONTROL

Optimization by mechanical calculations of the composite:

- put carbon fibers where it is needed
- orient the reinforcements in the direction of the loading.

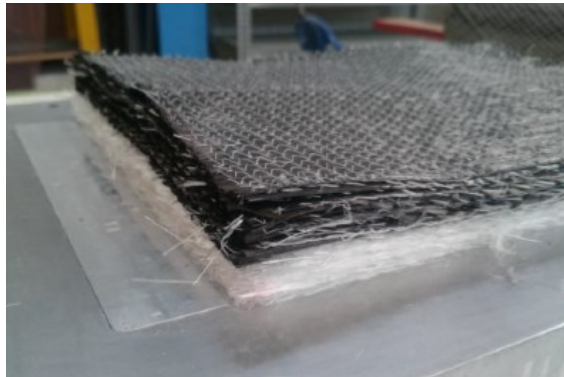
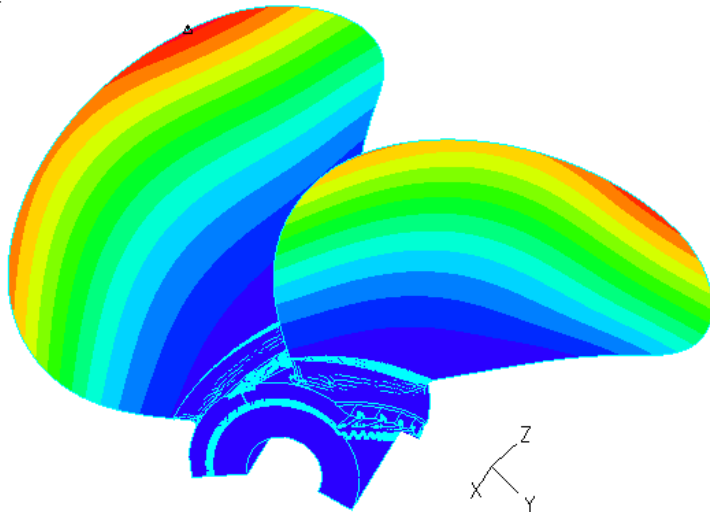
In connection with the manufacturing process.



DISPLACEMENT
MAGNITUDE
TIME 1.000

6.300
5.850
5.400
4.950
4.500
4.050
3.600
3.150
2.700
2.250
1.800
1.350
0.900
0.450
0.000

MAXIMUM
△ 5.998
NODE 53445
MINIMUM
* 0.000
NODE 121109





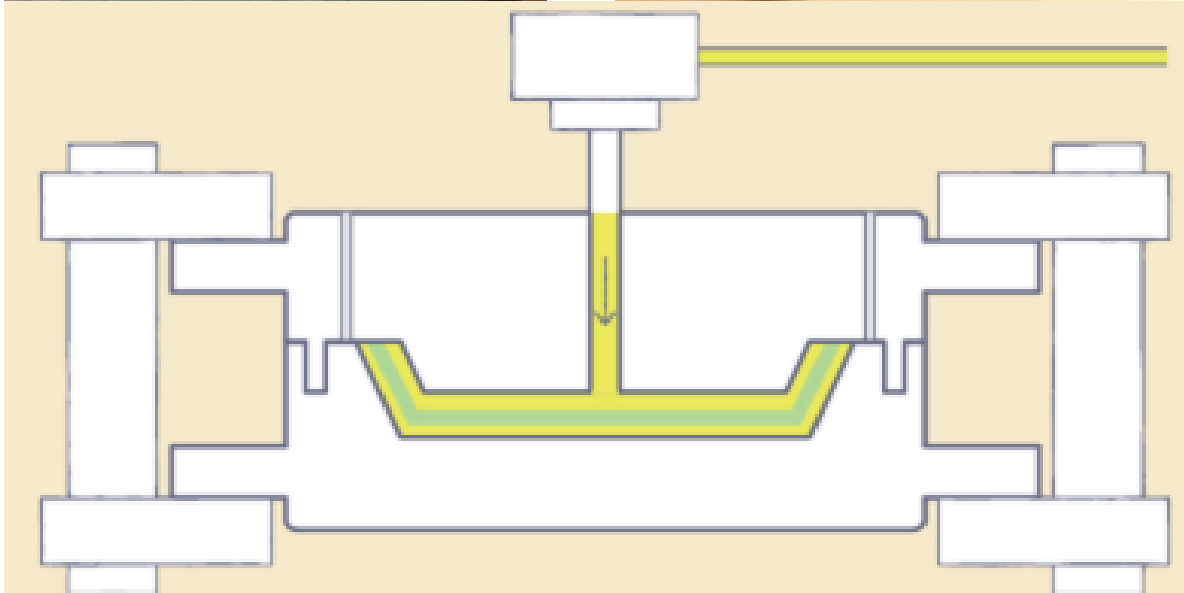
DEVELOPMENT OF THE MANUFACTURING PROCESS

Development of the manufacturing routing

Definition of process parameters and injection strategy

Process adjustments

Verification of process repeatability (deviation) and influencing parameters



RESIN TRANSFER MOLDING

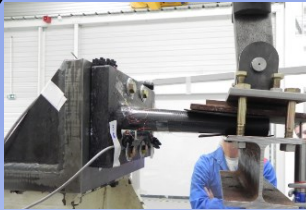
CONTROLS, TESTS

**SCALE
DEMONSTRATOR**

Tests on hub/blade connection mock-up

SCALE PROTOTYPE

Foils mock-up



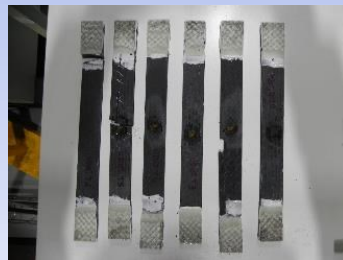
SCALE SAMPLES

Mechanical tests on samples

Samples, specimen and Material Controls and NDT, SEM, US

Impact tests on leading edges

Coating tests



TOOLS FOR MANUFACTURING PROCESS

Manufacture of blade tools

Fabrication of preforming tools



MOUNTING



SEA TRIALS





YOURS
QUESTIONS